

## Chapter 15

### CONCERNING THE FREE MOTION OF RIGID BODIES ACTED ON BY ANY FORCES.

---

#### THEOREM 10

**785.** In whatever way a rigid body may be acted on by forces, the momentary change is contained in these four effects : Initially by the variation of the speed of the centre of inertia, secondly by the change in the direction of the centre of inertia, thirdly by the variation of the angular speed about an axis of rotation passing through the centre of inertia, and fourthly by the variation in the axis of rotation itself.

#### DEMONSTRATION

In whatever manner a rigid body may be moving, the motion of this at some point of time can be resolved into a progressive motion in which the centre of inertia is moving, and a rotational motion about a certain axis passing through the centre of inertia ; from which the examination of this motion involves these four elements : 1<sup>st</sup> the speed of the centre of inertia; 2<sup>nd</sup> the direction along which it is moving; 3<sup>rd</sup> the axes passing through the centre of inertia, around which the body is rotating ; and 4<sup>th</sup> the rotational speed of this motion; having these four quantities known, the motion of the body present here becomes completely evident. But on account of the forces acting it possibly happens that these four quantities become changed, and thus an understanding of the effect of these is necessary, in order that we are able to define how much the individual effects may vary in an infinitely small element of time. Therefore the effect of the forces consists not only in these four outcomes, but in the momentary variation of these quantities, and if we are able to assign these then the effect becomes perfectly known to us; from which the truth of the theorem is evident.

#### COROLLARY 1

**786.** Therefore just as the effect of the forces on the motion of points is understood perfectly according to the change in the speed and direction; thus in the motion of rigid bodies it is necessary to know, in addition these two variations related to the centre of inertia, the variations undergone both by the axis of rotation as well as by the angular speed.

#### COROLLARY 2

**787.** Therefore in the same manner that we have defined the forces by which an acceleration is induced in the rotational motion about a fixed axis, thus also the forces are to be defined by which the above axis itself gains a given variation.

#### COROLLARY 3

**788.** Therefore the basis of the general theory of the motion of rigid bodies rests on this, that however the forces acting should be provided, we are able to assign these four variations produced in an element of the time.

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 786

**SCHOLIUM 1**

**789.** The principles leading to this conclusion have been set out well enough in the preceding work, where we have shown how it is necessary for a change to be determined in the motion of the centre of inertia as well as in the axis of rotation. Now because this later work, in which the main parts of this theory are contained, depends on the support of many investigations, which are often accustomed to involve many difficulties, here I propose thus that this theorem gathers these together as if into one, so that the problem can be solved by a single principle. Indeed I could have been using this easier method at the start, and thus I could not have avoided lightening the difficulties occurring in the above tract ; now in the argument presented so far so little has been examined that is not inconsistent with being very laborious and of the greatest prolixity, in which the individual ideas barely new in the mind are more firmly impressed, and these difficulties, with which this part of mechanics now seem to involve, are to be seen in a the clearest light. Now with nothing less than a new line of reasoning, I will set out this argument here as if anew, and neither will I call for support from anything carried forwards from before at this stage.

**SCHOLION 2**

**790.** Since the whole calculation may therefore be reduced here, it may be defined in order that the variations in the sizes of the four effects mentioned by the given forces are produced; because the direct method does not appear to be suitable for this, at first I may inquire instead about the forces, which are necessary to produce the given momentary variations, in order that we are able hence in turn to revert to that which we seek. And since there is no difficulty to be had in producing the variation in the motion of the centre of inertia, as I will observe that at rest; and I will investigate the forces required of this kind, so that the variations are taken of both the axis of rotation, about which the body is now turning, as well as the rotational speed acquired in an infinitely small time. For since the axis of rotation is assumed to be given an angular speed, the motion of the individual elements of the body have been given, which if resolved along three fixed directions, how much these three speeds change, both on account of the variation of the position of the axis of rotation, as well due to the variation of the angular speed, likewise can be deduced, and in the same way we are able to assign the forces producing these changes on the elements of the body ; and finally, from these elementary forces deduced, we will obtain the finite forces sought. Therefore since we must know the first motion of the individual elements of the body, while the body is rotating about some axis passing through the centre of inertia with a given angular speed, I will demonstrate the resolution of this along three fixed directions, which I take as the three principal axes of the body in the following problem.

**EULER'S**  
*Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.*

Chapter FIVETEEN.

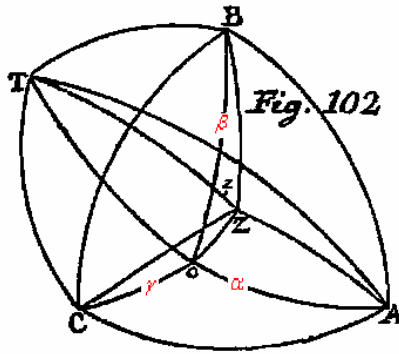
Translated and annotated by Ian Bruce.

page 787

**PROBLEMA 85**

**791.** If a rigid body should be rotating with a given speed about some axis passing through the centre of inertia, the motion of the individual elements of that is to be defined and that is to be resolved along the directions of the principal axes.

**SOLUTION**



About the centre of inertia  $I$ , which has not been expressed in the figure, a spherical surface is considered to be described (Fig. 102), in which  $A, B, C$  are the poles of the principal axes, thus in order that the arcs  $AB, AC$  and  $BC$  are quadrants. Now the body can rotate about some axis  $IO$  with an angular speed equal to  $\gamma'$  in the sense  $ABC$  and let the arcs for the pole of rotation  $O$  be  $OA = \alpha, OB = \beta$  and  $OC = \gamma$ . Now some element of the body is considered, from which a right line drawn to the centre of inertia  $I$  cuts the spherical surface at  $Z$ ; moreover let the distance of this from the centre of inertia  $I$  be equal

to  $r$ , while the radius of the sphere is put equal to one ; and it is evident that the motion of this element is similar to the motion of the point  $Z$ , while clearly the speed of this must be increased in the ratio 1 to  $r$ . Whereby it is sufficient for the motion of the point  $Z$  to be defined, if for which there is set up an arc  $ZzT$  normal to the arc  $OZ$ , then  $Zz$  is the direction of the motion and the speed is equal to  $\gamma' \sin OZ$ , because  $\sin OZ$  expresses the distance of the point  $Z$  from the axis of rotation  $IO$ . Moreover the quadrant arc  $ZT$  is put in place, in order that the radius  $IT$  is made parallel to the direction of the motion  $Zz$  [for in the limit,  $Zz$  acts along the tangent to the sphere], and now it is required to resolve the speed  $\gamma' \sin OZ$  produced along this direction  $IT$  in terms of the directions of the principal axes  $IA, IB, IC$ . Which at the end, with the arcs  $AT, BT, CT$  drawn which measure the inclination of that line  $IT$  to these principal axes, there is obtained

$$\text{the speed along } IA = \gamma' \sin OZ \cdot \cos AT ,$$

$$\text{the speed along } IB = \gamma' \sin OZ \cdot \cos BT$$

and

$$\text{the speed along } IC = \gamma' \sin OZ \cdot \cos CT .$$

Now since the arc  $OT$  is equally a quadrant, from the [spherical] triangle  $AOT$  there is found [from the cosine rule for sides :]

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 788

$$\cos AT = \cos AOT \cdot \sin AO = -\sin AOZ \cdot \sin AO$$

because  $TOZ = 90^\circ$ . In a similar manner,

$$\cos BT = \cos BOT \cdot \sin BO = \sin BOZ \cdot \sin BO$$

$$\cos CT = \cos COT \cdot \sin CO = \sin COZ \cdot \sin CO .$$

But on account of

$$\sin AZ : \sin AOZ = \sin OZ : \sin OAZ$$

then

$$\sin AOZ \cdot \sin OZ = \sin AZ \cdot \sin OAZ$$

and in a like manner

$$\sin BOZ \cdot \sin OZ = \sin BZ \cdot \sin OBZ$$

and

$$\sin COZ \cdot \sin OZ = \sin CZ \cdot \sin OCZ ;$$

from which there arises

$$\text{the speed along } IA = -\gamma' \sin AO \cdot \sin AZ \cdot \sin OAZ$$

$$\text{the speed along } IB = \gamma' \sin BO \cdot \sin BZ \cdot \sin OBZ$$

$$\text{the speed along } IC = \gamma' \sin CO \cdot \sin CZ \cdot \sin OCZ .$$

While now there arises [as  $BAC$  is a rt. angle, from the cosine rule for sides, and the quadrants, and then simply by multiplying and subtracting the ratios]

$$\sin BAO = \frac{\cos CO}{\sin AO}, \quad \cos BAO = \frac{\cos BO}{\sin AO}$$

$$\sin BAZ = \frac{\cos CZ}{\sin AZ}, \quad \cos BAZ = \frac{\cos BZ}{\sin AZ},$$

hence

$$\sin OAZ = \frac{\cos CO \cdot \cos BZ - \cos BO \cdot \cos CZ}{\sin AO \cdot \sin AZ}$$

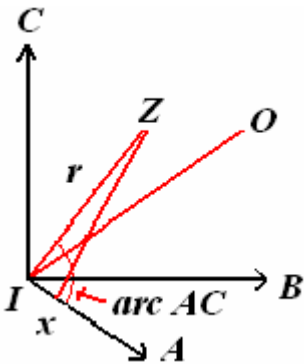
and thus

$$\text{the speed along } IA = \gamma'(\cos BO \cdot \cos CZ - \cos CO \cdot \cos BZ) ;$$

and in a like manner there is found,

$$\text{the speed along } IB = \gamma'(\cos CO \cdot \cos AZ - \cos AO \cdot \cos CZ) ,$$

$$\text{the speed along } IC = \gamma'(\cos AO \cdot \cos BZ - \cos BO \cdot \cos AZ) ,$$



which multiplied by  $r$  gives the speed of the proposed elements ;  
 from which if the coordinates  $x, y, z$  are put in place parallel to  
 the principal axes, then

$$r \cos AZ = x, r \cos BZ = y \text{ and } r \cos OZ = z ;$$

whereby since

$$AO = \alpha, BO = \beta, CO = \gamma$$

the speeds of the proposed elements are :

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 789

$$\begin{aligned} \text{speed along } IA &= \gamma'(z \cos \beta - y \cos \gamma) \\ \text{speed along } IB &= \gamma'(x \cos \gamma - z \cos \alpha) \\ \text{speed along } IC &= \gamma'(y \cos \alpha - x \cos \beta). \end{aligned}$$

**PROBLEM 86**

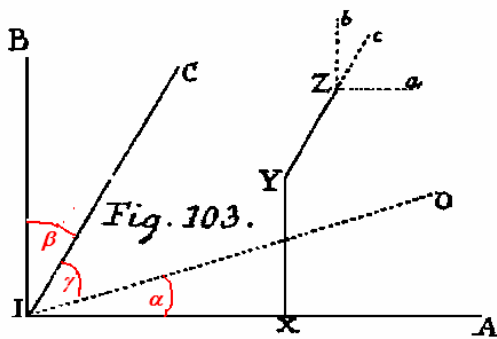
**792.** If a rigid body is rotating about some axis passing through the centre of inertia with a given angular speed, to find the elementary forces, on which the individual elements must be acted, in order that in the element of time  $dt$  both the axis of rotation itself as well as the angular speed undergo given changes.

**SOLUTION**

Let  $I$  be the centre of inertia of the body, the principal axes of which are  $IA, IB, IC$  and the body rotates about some axis  $IO$  (Fig. 103), the inclination of this to whichever axis shall be

$$AIO = \alpha, BIO = \beta, CIO = \gamma,$$

moreover the angular speed is equal to  $\gamma'$  following the sense  $ABC$ ; which quantities must



increase in the element of time  $dt$  by their differentials  $d\alpha, d\beta, d\gamma$  and  $d\gamma'$ , for which it is required to find the elementary forces producing this effect. Some element of the body  $dM$  must be considered situated at  $Z$ , for which the coordinates parallel to the principal axes are

$$IX = x, XY = y, YZ = z,$$

and the required forces effecting that prescribed motion and resolved along the principal axes may be called

$$Za = p, Zb = q \text{ and } Zc = r.$$

There is put in place the resolved [components of the] motion along the same directions  $Za = u$ , along  $Zb = v$  and along  $Zc = w$ , and since from first principles the motion shall be :

$$du = \frac{2gpdt}{dM}, \quad dv = \frac{2gqdt}{dM}, \quad dw = \frac{2grdt}{dM},$$

the forces sought become [recall that  $2g$  is a constant of proportionality, equal numerically to our acceleration of gravity]:

$$p = \frac{dudM}{2gdt}, \quad q = \frac{dvdM}{2gdt}, \quad r = \frac{dwdM}{2gdt}.$$

Now in the preceding problem we have thus found the three speeds  $u, v, w$  expressed in order that there becomes :

[This was the first occurrence of a cross product, which we would now write in vector notation as  $\vec{v} = \vec{\gamma}' \times \vec{r}$ ; Euler's great breakthrough was the discovery that the angular velocity could be expressed in components along the axes, and the components of the velocity can be expressed in terms of the cross product form shown; we have already met the dot product in problem 68 of Ch. 10. Thus, the laws for the two forms of vector multiplication were laid down by Euler in 1765.]

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 790

$$\begin{aligned} u &= \gamma'(z \cos \beta - y \cos \gamma) \\ v &= \gamma'(x \cos \gamma - z \cos \alpha) \\ w &= \gamma'(y \cos \alpha - x \cos \beta); \end{aligned}$$

which how much they are increased in the time  $dt$  both from the variation of the letters  $\gamma', d\alpha, d\beta, d\gamma$ , which are seen as given, while now it is necessary for the coordinates  $x, y, z$  to be judged. The differentials of these  $dx, dy, dz$  show the elements of the distances, through which the element  $dM$  is transferred in the time  $dt$ , thus in order that

$$\begin{aligned} dx &= udt = \gamma'(z \cos \beta - y \cos \gamma) \\ dy &= vdt = \gamma'(x \cos \gamma - z \cos \alpha) \\ dz &= wdt = \gamma'(y \cos \alpha - x \cos \beta); \end{aligned}$$

From which with differentiation put in place we duly arrive at :

$$\begin{aligned} du &= d\gamma'(z \cos \beta - y \cos \gamma) - \gamma'(z d\beta \sin \beta - y d\gamma \sin \gamma) + \gamma' dt(w \cos \beta - v \cos \gamma) \\ dv &= \gamma'(x \cos \gamma - z \cos \alpha) - \gamma'(x d\gamma \sin \gamma - z d\alpha \sin \alpha) + \gamma' dt(u \cos \gamma - w \cos \alpha) \\ dw &= \gamma'(y \cos \alpha - x \cos \beta) - \gamma'(y d\alpha \sin \alpha - x d\beta \sin \beta) + \gamma' dt(v \cos \alpha - u \cos \beta). \end{aligned}$$

Now since there is

$$\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1$$

and thus

$$\begin{aligned} \cos^2 \alpha + \cos^2 \beta &= \sin^2 \gamma, \\ \cos^2 \alpha + \cos^2 \gamma &= \sin^2 \beta \end{aligned}$$

and

$$\cos^2 \beta + \cos^2 \gamma = \sin^2 \alpha,$$

these formulas are changed into these :

$$\begin{aligned} du &= d\gamma'(z \cos \beta - y \cos \gamma) - \gamma' z d\beta \sin \beta + \gamma' y d\gamma \sin \gamma \\ &\quad + \gamma' \gamma' dt(y \cos \alpha \cos \beta + z \cos \gamma \cos \alpha - x \sin^2 \alpha) \\ dv &= \gamma'(x \cos \gamma - z \cos \alpha) - \gamma' x d\gamma \sin \gamma + \gamma' z d\alpha \sin \alpha \\ &\quad + \gamma' \gamma' dt(z \cos \beta \cos \gamma + x \cos \alpha \cos \beta - y \sin^2 \beta) \\ dw &= \gamma'(y \cos \alpha - x \cos \beta) - \gamma' y d\alpha \sin \alpha + \gamma' x d\beta \sin \beta \\ &\quad + \gamma' \gamma' dt(x \cos \gamma \cos \alpha + y \cos \beta \cos \gamma - y \sin^2 \gamma). \end{aligned}$$

Clearly these formulas are to be multiplied by  $\frac{dM}{2gd t}$ , from which the elementary forces sought  $p, q, r$  become known.

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 791

**COROLLARY 1**

**793.** Therefore if the individual elements of the body are acted on by three such forces, while the body is rotating about the axis  $IO$  with an angular speed  $\gamma'$ , in the lapse of time  $dt$  the angular speed  $\gamma'$  takes an increase equal to  $d\gamma'$  and likewise the axis of rotation thus changes about the principal axes  $IA, IB, IC$ , in order that the angles  $\alpha, \beta, \gamma$  are increased by their own differentials  $d\alpha, d\beta, d\gamma$ .

**COROLLARY 2**

**794.** In as much as we consider the forces acting on the same element of the body  $dM$ , the quantities  $x, y, z$  in these formulas are present as constants, because from these the position of the element is designated with respect to the principal axes, which always remain the same.

**COROLLARY 3**

**795.** But if we want to move from this element to others, and these forces acting are to be investigated, the same quantities  $x, y, z$  are variables and the remaining  $\alpha, \beta, \gamma, \gamma'$  with the differentials of these are to be regarded as constants, because these remain the same for all the elements of the body at the same instant.

**COROLLARY 4**

**796.** Whereby if we want to collect together all the elementary forces acting into one sum, only these formulas  $\int x dM, \int y dM$  and  $\int z dM$  occur to be integrated ; since the differentials of which [summed over] vanish, on account of the centre of inertia  $I$  of the body, and it is apparent that the sums of all the forces  $p$ , likewise  $q$  and  $r$  separately vanish.

**SCHOLIUM**

**797.** Because the sums of all the forces  $p, q$  and  $r$  vanish, which must always come about, as long as the centre of mass stays at rest in the same place, the effects of these is only to be decided from the moments of these ; and other forces also having the same moments produce the same effect, provided equal and opposite forces to these are applied to the centre of inertia. Now here it is not enough, that the forces have the same moment about one of several axes, for it is necessary that clearly all the axis produce the same moments, otherwise they cannot be taken as being equivalent. But this happens, provided the same moments are provided to the three principal axes, which is put beyond doubt in the following proposition.

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 792

**PROBLEM 87**

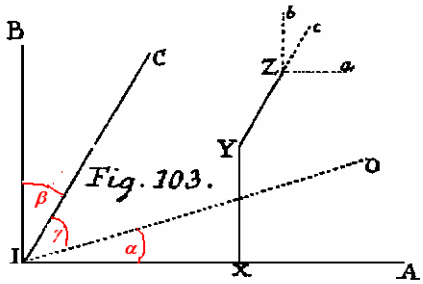
**798.** While a rigid body is rotating about some axis passing through the centre of inertia of the body corpus with a given angular speed, to define the moments of the forces about the principal axes, by which a change is induced both in the axis of rotation itself as well as in the given angular speed.

**SOLUTION**

With  $IO$  remaining for the axis of rotation with the angles

$$AIO = \alpha, BIO = \beta, CIO = \gamma,$$

(Fig. 103) about which the body now rotates with an angular speed equal to  $\gamma'$  in the sense  $ABC$ , and these quantities in the element of time  $dt$  must increase by their own differentials, some element of the body  $dM$  may be considered at  $Z$  with the coordinates



$$IX = x, XY = y \text{ et } YZ = z$$

with the elementary forces defined before :

$$Za = p = \frac{du \, dM}{2gdt}, \quad Zb = q = \frac{dv \, dM}{2gdt}, \quad Zc = r = \frac{dw \, dM}{2gdt};$$

from which the moment arises about the axis  $IA$  in the sense  $BC$

$$= ry - qz = \frac{dM}{2gdt} (ydw - zdv),$$

but about the axis  $IB$  the moment in the sense  $CA$

$$= pz - rx = \frac{dM}{2gdt} (zdu - xdw),$$

and finally about the axis  $IC$  the moment in the sense  $AB$

$$= qx - py = \frac{dM}{2gdt} (xdv - ydu).$$

But if here for  $du, dv, dw$  formulas found before are substituted, we find :

$$\begin{aligned} ydw - zdv &= d\gamma' \left( (yy + zz) \cos \alpha - xy \cos \beta - xz \cos \gamma \right) \\ &\quad - \gamma' (yy + zz) d\alpha \sin \alpha + \gamma' xy d\beta \sin \beta + \gamma' xz d\gamma \sin \gamma \\ &\quad + \gamma' \gamma' dt \left( (yy - zz) \cos \beta \cos \gamma + xy \cos \alpha \cos \gamma - xz \cos \alpha \cos \beta - yz (\sin^2 \gamma - \sin^2 \beta) \right) \\ zdu - xdw &= d\gamma' \left( (xx + zz) \cos \beta - yz \cos \gamma - xy \cos \alpha \right) \\ &\quad - \gamma' (xx + zz) d\beta \sin \beta + \gamma' yz d\gamma \sin \gamma + \gamma' xy d\alpha \sin \alpha \\ &\quad + \gamma' \gamma' dt \left( (zz - xx) \cos \alpha \cos \gamma + yz \cos \alpha \cos \beta - xy \cos \beta \cos \gamma - xz (\sin^2 \alpha - \sin^2 \gamma) \right) \end{aligned}$$



**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 793

$$\begin{aligned}
 xdv - ydu &= d\gamma'((xx + yy)\cos\gamma - xz\cos\alpha - yz\cos\beta) \\
 &- \gamma'(xx + yy)d\gamma\sin\gamma + \gamma'xz d\alpha\sin\alpha + \gamma'yz d\beta\sin\beta \\
 &+ \gamma'\gamma' dt \left( (xx - yy)\cos\alpha\cos\beta + xz\cos\beta\cos\gamma - yz\cos\alpha\cos\gamma - xy(\sin^2\beta - \sin^2\alpha) \right).
 \end{aligned}$$

Now these formulas can be multiplied by  $\frac{dM}{2gdt}$  and integrated through the whole mass of the body; then in the end  $Maa$ ,  $Mbb$ ,  $Mcc$  are the moments of inertia about the principal axes  $IA$ ,  $IB$ ,  $IC$  since, as there shall be

$$\begin{aligned}
 \int xxdM &= \frac{1}{2}M(bb + cc - aa), & \int yzdM &= 0, \\
 \int yydM &= \frac{1}{2}M(aa + cc - bb), & \int xzdM &= 0, \\
 \int zzdM &= \frac{1}{2}M(aa + bb - cc), & \int xydM &= 0,
 \end{aligned}$$

we obtain the three moments of the forces about the principal axes, from which the prescribed effect is produced, thus expressly :

1. The moment of the forces about the axis  $IA$  in the sense  $BC$

$$\frac{dM}{2gdt}(aad\gamma'\cos\alpha - \gamma'aad\alpha\sin\alpha + \gamma'\gamma'(cc - bb) dt \cos\beta\cos\gamma),$$

II. The moment of the forces about the axis  $IB$  in sense  $CA$

$$\frac{dM}{2gdt}(bbd\gamma'\cos\beta - \gamma'bbd\beta\sin\beta + \gamma'\gamma'(aa - cc) dt \cos\alpha\cos\gamma),$$

III. The moment of the forces about the axis  $IC$  in sense  $AB$

$$\frac{dM}{2gdt}(ccd\gamma'\cos\gamma - \gamma'ccd\gamma\sin\gamma + \gamma'\gamma'(bb - aa) dt \cos\alpha\cos\beta).$$

**COROLLARY I**

**799.** Thus in order that the body rotates uniformly about the same axis, the three moments of the forces on account of

$$d\gamma' = 0, d\alpha = 0, d\beta = 0, d\gamma = 0$$

become

$$\begin{aligned}
 \text{I.} &= \frac{M\gamma'\gamma'(cc - bb)\cos\beta\cos\gamma}{2g}, \\
 \text{II.} &= \frac{M\gamma'\gamma'(aa - cc)\cos\alpha\cos\gamma}{2g}, \text{ and} \\
 \text{III.} &= \frac{M\gamma'\gamma'(bb - aa)\cos\alpha\cos\beta}{2g},
 \end{aligned}$$

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
**Chapter FIVETEEN.**

Translated and annotated by Ian Bruce.

page 794

which do not vanish, unless the axis of rotation coincides with one on the principal axes.

**COROLLARY 2**

**800.** It is understood in a similar manner, which forces are required, in order that either the angular speed alone is changed or the position of the axis of rotation is varied ; clearly the forces, the moments of which agree with those defined before, are outstanding in this, but only if equal and opposite forces to these are applied to the centre of inertia, in order that these forces are able to be taken as vanishing and the total effect of these must be in their moment alone.

**COROLLARY 3**

**801.** If the body is rotating about some principal axis  $IA$  with an angular speed  $\gamma'$ , which must be increased by its own differential  $d\gamma'$ , on account of  $\alpha = 0$  and  $\beta = \gamma = 90^\circ$  according to this is there only a moment of the force equal to  $\frac{dMaad\gamma'}{2gdt}$  required about the axis  $IA$ , as we have now found above.

**SCHOLIUM**

**802.** This problem would have been no more difficult to solve, if in addition to the rotational motion above we should attribute some progressive motion, which in the time element  $dt$  also must be varied in a prescribed manner ; if indeed the centre of inertia has some motion which resolved along the principal axes present the speeds  $l, m, n$ , and in the element of time  $dt$  are to be increased by their own differentials, the above values of the speeds  $u, v, w$  arising from the rotational motion must be augmented by these progressive speeds  $l, m, n$ , and from the increments of these forces arise, the equivalent of which pass through the centre of inertia and may be obtained equally, as if the body were able to pursue this motion without any rotational motion. From which that can be confirmed, which now we have shown above, that in such a mixed motion the progressive motion and the rotational motion can always be separated and each in turn, as if the other were not present, clearly can be considered and determined.

**PROBLEM 88**

**803.** If a rigid body, while rotating about a given axis  $IO$  with a given angular acceleration equal to  $\gamma'$ , should be acted on by some forces, to which likewise equal and opposite forces are applied to the centre of inertia itself, to determine both the change of the axis, as well as the change of the angular speed produced in the time  $dt$ .

**SOLUTION**

The moments of the forces are gathered together about the three principal axes of the body and let

the moment of the forces about the axis  $IA$  in the sense  $BC = P$ ,  
the moment of the forces about the axis  $IB$  in the sense  $CA = Q$ ,  
the moment of the forces about the axis  $IC$  in the sense  $AB = R$ .

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 795

Moreover the moments of inertia of the body about the same axes are thus far  $Maa$ ,  $Mbb$ ,  $Mcc$ .  
 But if now the body is rotating in the sense  $ABC$  with an angular speed equal to  $\gamma'$  about the axis  $IO$ , the inclinations to the same principal axes are

$$AIO = \alpha, BIO = \beta, CIO = \gamma,$$

these quantities undergo the following changes in the time  $dt$ :

$$\begin{aligned} \frac{2gPdt}{Maa} &= d\gamma' \cos \alpha - \gamma' d\alpha \sin \alpha + \frac{cc-bb}{aa} \gamma' \gamma' dt \cos \beta \cos \gamma \\ \frac{2gPdt}{Mbb} &= d\gamma' \cos \beta - \gamma' d\beta \sin \beta + \frac{aa-cc}{bb} \gamma' \gamma' dt \cos \alpha \cos \gamma \\ \frac{2gPdt}{Mcc} &= d\gamma' \cos \gamma - \gamma' d\gamma \sin \gamma + \frac{bb-aa}{cc} \gamma' \gamma' dt \cos \alpha \cos \beta, \end{aligned}$$

from which equations the four unknowns  $\alpha$ ,  $\beta$ ,  $\gamma$ , et  $\gamma'$  are to be determined, which are to be had from three equations only on account of

$$\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1.$$

Therefore since there becomes

$$d\alpha \sin \alpha \cos \alpha + d\beta \sin \beta \cos \beta + d\gamma \sin \gamma \cos \gamma = 0,$$

if the first equation is multiplied by  $\cos \alpha$ , the second by  $\cos \beta$ , and the third by  $\cos \gamma$  there is produced on the products being added :

$$\begin{aligned} d\gamma' + \left( \frac{cc-bb}{aa} + \frac{aa-cc}{bb} + \frac{bb-aa}{cc} \right) \gamma' \gamma' dt \cos \alpha \cos \beta \cos \gamma \\ = \frac{2gdt}{M} \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right) \end{aligned}$$

or

$$\begin{aligned} d\gamma' &= \frac{(cc-bb)(aa-cc)(bb-aa)}{aabbcc} \gamma' \gamma' dt \cos \alpha \cos \beta \cos \gamma \\ &+ \frac{2gdt}{M} \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right), \end{aligned}$$

with which value substituted these equations are obtained :

## EULER'S

### *Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.*

#### Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 796

$$\begin{aligned} \gamma' d\alpha \sin \alpha &= \frac{(cc-bb)}{aa} \gamma' \gamma' dt \cos \beta \cos \gamma \left( 1 + \frac{(aa-cc)(bb-aa)}{bbcc} \cos^2 \alpha \right) \\ &\quad + \frac{2gdt}{M} \left( \frac{Q \cos \alpha \cos \beta}{bb} + \frac{R \cos \alpha \cos \gamma}{cc} - \frac{P \sin^2 \alpha}{aa} \right) \\ \gamma' d\beta \sin \beta &= \frac{(aa-cc)}{bb} \gamma' \gamma' dt \cos \alpha \cos \gamma \left( 1 + \frac{(bb-aa)(cc-bb)}{aacc} \cos^2 \beta \right) \\ &\quad + \frac{2gdt}{M} \left( \frac{R \cos \beta \cos \gamma}{cc} + \frac{P \cos \beta \cos \alpha}{aa} - \frac{Q \sin^2 \beta}{bb} \right) \\ \gamma' d\gamma \sin \gamma &= \frac{(bb-aa)}{cc} \gamma' \gamma' dt \cos \alpha \cos \beta \left( 1 + \frac{(cc-bb)(aa-cc)}{aabb} \cos^2 \gamma \right) \\ &\quad + \frac{2gdt}{M} \left( \frac{P \cos \gamma \cos \alpha}{aa} + \frac{Q \cos \gamma \cos \beta}{bb} - \frac{R \sin^2 \gamma}{cc} \right). \end{aligned}$$

But if the first of these equations is multiplied by  $aa \cos \alpha$ , the second by  $bb \cos \beta$ , and the third by  $cc \cos \gamma$ , these on being added give rise to

$$\begin{aligned} \frac{2gdt}{M} (P \cos \alpha + Q \cos \beta + R \cos \gamma) &= d\gamma' \left( aa \cos^2 \alpha + bb \cos^2 \beta + cc \cos^2 \gamma \right) \\ &\quad - \gamma' (aad\alpha \sin \alpha \cos \alpha + bbd\beta \sin \beta \cos \beta + ccd\gamma \sin \gamma \cos \gamma), \end{aligned}$$

which multiplied by  $2M\gamma'$  and with the other part integrated gives

$$M\gamma' \gamma' \left( aa \cos^2 \alpha + bb \cos^2 \beta + cc \cos^2 \gamma \right) = 4g \int \gamma' dt (P \cos \alpha + Q \cos \beta + R \cos \gamma),$$

which quantity expresses the *vis viva* of the body.

### COROLLARY 1

**804.** If therefore, while a body is rotating about some axis passing through the centre of inertia, it is acted on by some forces, hence the momentary variations both in the axis of rotation in place as well as in the angular speed are determined.

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 797

**COROLLARY 2**

**805.** If the body is not acted on by any external forces, then the axis of rotation as well as the angular speed, thus are varied in order that :

$$\begin{aligned} \text{I. } d\gamma' &= \frac{(cc-bb)(aa-cc)(bb-aa)}{aabbcc} \gamma' \gamma' dt \cos \alpha \cos \beta \cos \gamma \\ \text{II. } d\alpha \sin \alpha &= \frac{cc-bb}{aa} \gamma' dt \cos \beta \cos \gamma \left( 1 + \frac{(aa-cc)(bb-aa)}{bbcc} \cos^2 \alpha \right) \\ \text{III. } d\beta \sin \beta &= \frac{aa-cc}{bb} \gamma' dt \cos \alpha \cos \gamma \left( 1 + \frac{(bb-aa)(cc-bb)}{aacc} \cos^2 \beta \right) \\ \text{IV. } d\gamma \sin \gamma &= \frac{bb-aa}{cc} \gamma' dt \cos \alpha \cos \beta \left( 1 + \frac{(cc-bb)(aa-cc)}{aabb} \cos^2 \gamma \right) \end{aligned}$$

and the *vis viva*  $M\gamma'\gamma'(aa \cos^2 \alpha + bb \cos^2 \beta + cc \cos^2 \gamma)$  always remains constant.

**COROLLARY 3**

**806.** If the body should be at rest, in order that  $\gamma' = 0$ , from the moments of the forces  $P$ ,  $Q$ ,  $R$  taken about the principal axed, the axis, about which the body begins to rotate, can be defined from these equations

:

$$\frac{Q \cos \alpha \cos \beta}{bb} + \frac{R \cos \alpha \cos \gamma}{cc} - \frac{P \sin^2 \alpha}{aa} = 0$$

or

$$\begin{aligned} \frac{P}{aa} &= \cos \alpha \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right), \\ \frac{R \cos \beta \cos \gamma}{cc} + \frac{P \cos \beta \cos \alpha}{aa} - \frac{Q \sin^2 \beta}{bb} &= 0 \end{aligned}$$

or

$$\begin{aligned} \frac{Q}{bb} &= \cos \beta \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right), \\ \frac{P \cos \gamma \cos \alpha}{aa} + \frac{Q \cos \gamma \cos \beta}{bb} - \frac{R \sin^2 \gamma}{cc} &= 0 \end{aligned}$$

or

$$\frac{R}{cc} = \cos \gamma \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right),$$

from which, since

$$\cos \alpha : \cos \beta : \cos \gamma = \frac{P}{aa} : \frac{Q}{bb} : \frac{R}{cc},$$

then

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 798

$$\cos \alpha = \frac{P}{aa} \cdot \sqrt{\left(\frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4}\right)}$$

$$\cos \beta = \frac{Q}{bb} \cdot \sqrt{\left(\frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4}\right)}$$

$$\cos \gamma = \frac{R}{cc} \cdot \sqrt{\left(\frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4}\right)},$$

and in the element of time  $dt$  there is produced

$$d\gamma' = \frac{2gdt}{M} \sqrt{\left(\frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4}\right)}.$$

**SCHOLIUM 1**

**807.** Therefore in this problem alone, everything is contained which we elicited above with great labour by many circuitous routes [§ 639], yet since here we can use the motion from first principles and everything becomes most transparent. Thus that above, while the body is at rest, the axis, about which the forces of the first rotation are impressed, we were able to determine only with the greatest labour, here this equivalent determination of the corollary can be derived at once from the present problem; the agreement of this with the above to which it is evidently easier, and it may be done without the foolish ambiguity of the square root sign, if again for the axis of rotation  $IF$  (Fig. 82, ) the angle  $AIE = \eta$  and the angle  $EIF = \vartheta$ , then

$$\cos \alpha = \cos \eta \cos \vartheta,$$

$$\cos \beta = -\sin \eta \cos \vartheta$$

and

$$\cos \gamma = \sin \vartheta,$$

from which on account of

$$\text{tang } \eta = \frac{-\cos \beta}{\cos \alpha}$$

there is

$$\text{tang } \eta = \frac{-Qaa}{Pbb}$$

and

$$\text{tang } \vartheta = \frac{\cos \gamma}{\cos \alpha} \cos \eta = \frac{Raa}{Pcc} \cos \eta.$$

Moreover since the forces acting here shall be

$$VP = P, VQ = Q, VR = R$$

with the angle present  $AIV = \delta$  and  $IV = h$ , then the moment of the forces about the axis  $IA$  in the sense  $BC$  is equal to  $Rh \sin \delta$ , which is for us here  $P$ ; then the moment of these about the axis  $IB$  in the sense  $CA$  is equal to  $-Rh \cos \delta$ , which is for us here  $Q$ , and the moment about the axis  $IC$  in the sense  $AB$  is equal to

$$Qh \cos \delta - Ph \sin \delta,$$

which is for us here  $R$ . With which values for  $P$ ,  $Q$  and  $R$  put in place we have again in short

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 799

$$\text{tang } \eta = \frac{aa \cos \delta}{bb \sin \delta}$$

and

$$\text{tang } \vartheta = \frac{aa(Q \cos \delta - P \sin \delta)}{ccR \sin \delta} \cos \eta.$$

Then also, that which finally we have elicited above from the momentary change in the rotational motion, while the body is rotating about a non-principal axis with no external forces acting, through exceedingly intricate reasoning, here with the moments of the forces put as  $P = 0$ ,  $Q = 0$ ,  $R = 0$  these become the most evident, as we have shown in corollary 2. But that which we hardly dared to touch on above, with the above body acted on by any forces, with equal facility and the same labour here happily we have set out, thus in order that only in this chapter have we been seen to put in place fully a general theorem for the motion of rigid bodies established from the first principles of motion.

**SCHOLIUM 2**

**808.** But since with any proposed forces acting, of which the moments  $P$ ,  $Q$ ,  $R$  shall be taken about the principal axes in the sense  $ABC$ , the whole calculation turns on the determination of the three angles  $\alpha$ ,  $\beta$ ,  $\gamma$  and of the speed  $\gamma'$ , for which we have found three equations, seeing that these angles maintain a relation between each other, these equations by a trifling substitution are able to be reduced much more conveniently. For since we do not need these letters  $x$ ,  $y$ ,  $z$  to further indicate the nature of the body, if we put

$$\gamma' \cos \alpha = x, \quad \gamma' \cos \beta = y, \quad \gamma' \cos \gamma = z,$$

then all the angles are cancelled out of from the calculation and the sum of the whole theory of the motion of rigid bodies is contained in these three formulas simply enough :

$$\begin{aligned} dx + \frac{cc-bb}{aa} yzdt &= \frac{2gPdt}{Maa} \\ dy + \frac{aa-cc}{bb} zxdt &= \frac{2gQdt}{Mbb} \\ dz + \frac{bb-aa}{cc} xydt &= \frac{2gRdt}{Mcc}. \end{aligned}$$

Whereby if the body is acted on by no forces, we deduce at once

$$aaxdx + bbydy + cczdz = 0$$

or

$$aaxx + bbyy + cczz = \text{Const.}$$

Then from two equations on  $dt$  being removed, there will be

$$\frac{aadx}{bbdy} = \frac{(cc-bb)y}{(aa-cc)x}$$

and thus on integrating

$$\frac{aa}{cc-bb} xx = \frac{bb}{aa-cc} yy + \text{Const.}$$

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 800

Whereby if initially there was

$$x = \mathfrak{A}, \quad y = \mathfrak{B}, \quad z = \mathfrak{C}$$

and we put

$$\frac{aa}{cc-bb} = A, \quad \frac{bb}{aa-cc} = B \quad et \quad \frac{cc}{bb-aa} = C,$$

then there is had

$$Axx - Byy = A\mathfrak{A}^2 - B\mathfrak{B}^2$$

and

$$Axx - Cz z = A\mathfrak{A}^2 - C\mathfrak{C}^2$$

and thus

$$y = \frac{\sqrt{(Axx - A\mathfrak{A}^2 + B\mathfrak{B}^2)}}{\sqrt{B}}$$

and

$$z = \frac{\sqrt{(Axx - A\mathfrak{A}^2 + C\mathfrak{C}^2)}}{\sqrt{C}}.$$

Whereby since

$$Adx + yzdt = 0,$$

there becomes

$$dt = - \frac{Adx\sqrt{BC}}{\sqrt{(Axx - A\mathfrak{A}^2 + B\mathfrak{B}^2)}\sqrt{(Axx - A\mathfrak{A}^2 + C\mathfrak{C}^2)}}$$

and thus the solution of this problem, that above gave us considerable difficulty, has been established in a satisfactory manner.



**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

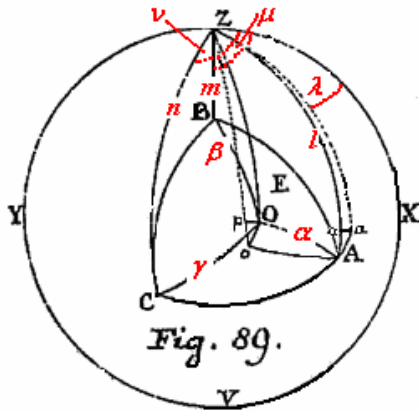
page 801

**PROBLEM 89**

**809.** If we should know the axis of rotation relative to the principal axes at some time together with the angular speed of the body about this axis, then at some time to define the position of the principal axes with respect to absolute space.

**SOLUTION**

A motionless sphere is considered in absolute space, at the centre of this sphere the centre of inertia  $I$  of the body is turning (Fig. 89) and on that sphere there is taken a fixed great circle  $ZXVY$ , and on that in turn a fixed point  $Z$ , to which the position of the principal axes are referred at any time. And now indeed in the elapsed time  $t$  the principal axes of the body correspond to the fixed points on the sphere  $A, B, C$ , if from which to  $Z$  the arcs of the great circles are drawn, these can be called



$ZA = l, ZB = m$  and  $ZC = n$ ,  
 then the angles will be  
 $XZA = \lambda, XZB = \mu$  et  $XZC = \nu$ .

But now the axis of rotation is found at  $O$ , in order that

$$AO = \alpha, BO = \beta \text{ et } CO = \gamma,$$

about which the body is rotating in the sense  $ABC$  with the angular speed  $\gamma'$ ; hence in the element of time  $dt$  the pole  $A$  is turned through the arclet

$$Aa = \gamma' dt \sin \alpha$$

with  $Aa$  present normal to the arc  $OA$ , thus in order that

$$\sin BAa = \frac{\cos \beta}{\sin \alpha} \quad \text{and} \quad \cos BAa = \frac{\cos \gamma}{\sin \alpha}.$$

But it is the case that

$$\sin ZAB = -\frac{\cos n}{\sin l} \quad \text{and} \quad \cos ZAB = \frac{\cos m}{\sin l},$$

from which it may be deduced

$$\sin ZAa = \frac{\cos \beta \cos m + \cos \gamma \cos n}{\sin \alpha \sin l},$$

$$\cos ZAa = \frac{\cos \gamma \cos m - \cos \beta \cos n}{\sin \alpha \sin l}.$$

Now on drawing the perpendicular  $a\alpha$  from  $a$  to the arc  $ZA$ , then

$$A\alpha = \frac{\gamma' dt}{\sin l} (\cos \gamma \cos m - \cos \beta \cos n)$$

and

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 802

$$a\alpha = \frac{\gamma' dt}{\sin l} (\cos \beta \cos m + \cos \gamma \cos n).$$

Then indeed

$$A\alpha = -dl \quad \text{and} \quad a\alpha = -d\lambda \sin l$$

and thus hence on account of the ratio, the values of the following differentials can be established:

$$\begin{aligned} dl \sin l &= \gamma' dt (\cos \beta \cos n - \cos \gamma \cos m) \\ d\lambda \sin^2 l &= -\gamma' dt (\cos \beta \cos m + \cos \gamma \cos n) \\ dm \sin m &= \gamma' dt (\cos \gamma \cos l - \cos \alpha \cos n) \\ d\mu \sin^2 m &= -\gamma' dt (\cos \gamma \cos n + \cos \alpha \cos l) \\ dn \sin n &= \gamma' dt (\cos \alpha \cos m - \cos \beta \cos l) \\ dv \sin^2 n &= -\gamma' dt (\cos \alpha \cos l + \cos \alpha \cos m). \end{aligned}$$

It will suffice to resolve the first two of these three, since

$$\cos^2 l + \cos^2 m + \cos^2 n = 1,$$

and with these resolved with the single or the remaining, the whole calculation is complete.

**COROLLARY 1**

**810.** If we put

$$\gamma' \cos \alpha = x, \quad \gamma' \cos \beta = y \quad \text{and} \quad \gamma' \cos \gamma = z,$$

in order that from the moments of the forces acting  $P, Q, R$  there becomes :

$$\begin{aligned} dx + \frac{cc - bb}{aa} yz dt &= \frac{2gPdt}{Maa}, \\ dy + \frac{aa - cc}{bb} xz dt &= \frac{2gQdt}{Mbb}, \\ dz + \frac{bb - aa}{cc} xy dt &= \frac{2gRdt}{Mcc}, \end{aligned}$$

now it is required to add the following equations :

$$\begin{aligned} dl \sin l &= dt (y \cos n - z \sin m) \\ dm \sin m &= dt (z \cos l - x \cos n) \\ dn \sin n &= dt (x \cos m - y \cos l) \\ d\lambda \sin^2 l &= -dt (y \cos m + z \cos n) \\ d\mu \sin^2 m &= -dt (z \cos n + x \cos l) \\ dv \sin^2 n &= -dt (x \cos l + y \cos m). \end{aligned}$$

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 803

**COROLLARY 2**

**811.** If again we put

$$\cos l = p, \cos m = q, \cos n = r,$$

the latter equations adapt these forms on account of  $pp + qq + rr = 1$ :

$$dp + dt(yr - zq) = 0$$

$$dq + dt(zp - xr) = 0$$

$$dr + dt(xq - yp) = 0$$

$$d\lambda + \frac{dt(yq + zr)}{qq + rr} = 0$$

$$d\mu + \frac{dt(zr + xp)}{pp + rr} = 0$$

$$d\nu + \frac{dt(xp + yq)}{pp + qq} = 0,$$

thus also there shall be

$$xdp + ydq + zdr = 0,$$

just as

$$pdp + qdq + rdr = 0.$$

**SCHOLIUM**

**812.** And if I have considered the preceding problem here, just as if it were now solved, yet generally both problems are joined together and it is necessary to put in place the solution of these likewise, just as comes about in the use in the preceding chapter on the motion of tops. Clearly it is necessary that both problems are to be joined together, when the forces acting depend on the absolute position of the body, which indeed, if external forces might be present, usually always happens. Therefore from these cases the moments of the forces  $P, Q, R$ , the arcs  $l, m, n$  and perhaps also the angles  $\lambda, \mu, \nu$  may be involved, thus in order that all the equations in Corollary 1 must be shown to be assessed likewise, before a solution can be accepted. But if the above body should be carried forwards by a progressive motion, it is accustomed to happen, that the forces also depend on that, from which it is necessary that the formulas involving progressive motion likewise are to be added to the others, in which cases a very complicated solution may be returned. Therefore now from these problems set out we are able to undertake the general problem concerning the free motion of rigid bodies acted on by any forces.

**PROBLEM 90**

**813.** If a rigid body projected initially in some manner and then is acted on by forces of some kind, to the action of which it can freely comply, then to determine the motion of this body.

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 804

**SOLUTION**

Because in the first place according to the progressive motion, or the motion by which the centre of inertia is moved forwards, this can be defined by the same precepts which have been laid down for the motion of points. Clearly the whole mass of the body, which is equal to  $M$ , is considered to be gathered together at the centre of inertia of this, and to be acted on by the forces, therefore the motion of this can be determined from the precepts laid down above or in any case can be expressed by analytical formulas, without taking into account the rotational motion, which meanwhile perhaps the body is performing about the centre of inertia. Then truly in order that this motion can be investigated, in the first place with the progressive motion thoroughly set apart, the centre of inertia now can be considered as at rest; and indeed initially the three principal axes are to be investigated, which shall be  $IA$ ,  $IB$ ,  $IC$ , drawn from the centre of inertia  $I$  and the moments of inertia of these about the principal axes shall be  $Maa$ ,  $Mbb$ ,  $Mcc$ ; from which known a stationary sphere is considered to be described about the centre of inertia  $I$ , on which a great circle  $ZXVY$  is taken as well as a fixed point  $Z$  on that, to which the position of the body at some time may be referred. Now therefore in the elapsed time equal to  $t$  the body maintains a position on account of the rotational motion represented in the figure, in which the principal axes correspond to the points  $A$ ,  $B$ ,  $C$  on the spherical surface with the points distant by a quadrant from each other in turn, from the present position of which there can be put

$$\text{the arc } ZA = l, \quad ZB = m \text{ and } ZC = n$$

likewise

$$\text{the angles } XZA = \lambda, \quad XZB = \mu, \quad XZC = \nu,$$

and how they depend on each other in turn, is evident from the sphere. Again the body now is rotating about the axis  $IO$  with an angular speed  $\gamma'$  in the sense  $ABC$ , and for the position of this axis there are put in place the arcs

$$AD = \alpha, \quad BD = \beta, \quad CO = \gamma;$$

and these are quantities thus to be determined from their own differentials, in order that on putting  $t = 0$  they agree to the initial position of the body. To this the forces acting on the body now can be considered, of which the moments of inertia about the principal axes can be deduced, and let

$$\text{the moment of the forces about the axis } IA \text{ in the sense } BC = P,$$

$$\text{the moment of the forces about the axis } IB \text{ in the sense } CA = Q,$$

$$\text{the moment of the forces about the axis } IC \text{ in the sense } AB = R,$$

and on putting, for the sake of brevity

$$\gamma' \cos \alpha = x, \quad \gamma' \cos \beta = y \quad \text{and} \quad \gamma' \cos \gamma = z,$$

in order that

$$\gamma' = \sqrt{(x^2 + y^2 + z^2)},$$

we have found above to become:

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 805

$$\begin{aligned} dx + \frac{cc - bb}{aa} yz dt &= \frac{2gPdt}{Maa}, \\ dy + \frac{aa - cc}{bb} xz dt &= \frac{2gQdt}{Mbb}, \\ dz + \frac{bb - aa}{cc} xy dt &= \frac{2gRdt}{Mcc}, \end{aligned}$$

[We can compare these results, which are called Euler's Equations of Motion, with those in modern texts on dynamics, in a notation that is hopefully obvious to the reader : e. g. *Principles of Mechanics*, p. 354 § 354 Synge & Griffith, [1<sup>st</sup> Ed. (1942) McGraw-Hill], Motion of a rigid body :

$$\begin{aligned} A\dot{\omega}_1 - (B - C)\omega_2\omega_3 &= G_1; \\ B\dot{\omega}_2 - (C - A)\omega_3\omega_1 &= G_2; \\ C\dot{\omega}_3 - (A - B)\omega_1\omega_2 &= G_3; \end{aligned}$$

where  $2gP$ , etc are the moments about the principal axes in Euler's original equations corresponding to  $G_1$ , etc.]  
 since on putting

$$\cos l = p, \quad \cos m = q \quad \text{et} \quad \cos n = r$$

these equations arise:

$$\begin{aligned} dp + dt(yr - zq) &= 0 \\ dq + dt(zp - xr) &= 0 \\ dr + dt(xq - yp) &= 0 \\ d\lambda + \frac{dt(yq + zr)}{qq + rr} &= 0 \\ d\mu + \frac{dt(zr + xp)}{rr + pp} &= 0 \\ dv + \frac{dt(xp + yq)}{pp + qq} &= 0, \end{aligned}$$

which if they are thus to be resolved and integrated, so that at some time  $t$  they are able to be assigned the quantities  $x, y, z, p, q, r, \lambda, \mu, v$ , the problem becomes completely solved.

Moreover from these final equations it is to be noted that

$$pp + qq + rr = 1,$$

from which

$$pdp + qdq + rdr = 0,$$

then also

$$xdp + ydq + zdr = 0.$$

Finally

$$\sin(\mu - \lambda) = -\frac{\cos n}{\sin l \sin m}$$

and

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***

*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 806

$$\cos(\mu - \lambda) = -\frac{\cos l \cos m}{\sin l \sin m},$$

and thus

$$\text{tang}(\mu - \lambda) = \frac{r}{pq}$$

and

$$\text{tang}(\nu - \mu) = \frac{p}{qr},$$

while

$$\text{tang}(\lambda - \nu) = \frac{q}{pr}$$

or

$$\text{tang}(\nu - \lambda) = -\frac{q}{pr},$$

thus so that it suffices that one of the angles  $\lambda$ ,  $\mu$ ,  $\nu$  to be found.

**SCHOLIUM**

**814.** These precepts for the determination of the motion of rigid bodies appear to be the most general, nor are they restricted to free motion ; for in whatever way the motion of these is restrained, either upon a certain plane or they are known to advance towards other bodies, or some point of these if kept fixed, the question can always be reduced to the rules established. Clearly when contact is made with part of another body, there a force is given, which initially is introduced into the calculation as an unknown and must hence be determined next, in order that the motion can be returned in agreement with the proposed conditions ; and thus in this way the impact between bodies can be examined. Before we undertake investigations of this kind, it will be convenient for some time to be expended on a certain case of the free motion of a body, in which the rotational motion about the position of a variable axis may be found, while the body is acted on by external forces ; motion of this kind is not produced by the force of gravity, clearly the direction of this force passes through the centre of inertia of the body. But the most serious question of this kind of motion without doubt lies in the rotational motion of celestial bodies, but which cannot be undertaken without the theoretical principles of astronomy being in place. But from the consensus of all the observations, which at this point it is allowed to put in place, it has been ascertained that celestial bodies are likewise moving, and they should either attract or in turn repel each other by mutual forces, if these are in the reciprocal ratio of the squares of the distances and proportional to the above masses. Clearly just as some heavy bodies are turned towards the earth, thus also they have a certain force exerted on them towards all celestial bodies, which avoids being larger than that of the earth because these forces are diminished more by the square of the distance. And from these forces astronomers are accustomed to scrutinize the progress of celestial bodies, which investigation can be referred to as the motion of points, yet here we inquire into the rotational motion of celestial bodies, which argument I will thus strive to set out generally in the following chapter, thus so that astronomy shall soon be advanced in the following and not ignored.

## CAPUT XV

### **DE MOTU LIBERO CORPORUM RIGIDORUM A VIRIBUS QUIBUSCUNQUE SOLLICITATORUM**

---

#### THEOREMA 10

**785.** Quomodocunque corpus rigidum a viribus sollicitetur, effectus momentaneus his quatuor rebus continetur: primo variatione celeritatis centri inertiae, secundo variatione directionis centri inertiae, tertio variatione celeritatis angularis circa axem gyrationis per centrum inertiae transeuntis, et quarto variatione ipsius axis gyrationis.

#### DEMONSTRATIO

Quomodocunque corpus rigidum moveatur, eius motus quovis temporis puncto resolvitur in motum progressivum, quo centrum inertiae movetur, et motum gyratorium circa axem quempiam per centrum inertiae transeuntem; unde cognitio huius motus haec quatuor elementa involvit: 1° celeritatem centri inertiae, 2° directionem, secundum quam movetur, 3° axem per centrum inertiae transeuntem, circa quem corpus iam gyrationis et 4° celeritatem angularem huius motus; quas quatuor res qui cognoverit, motum corporis hoc instanti perfecte habet perspectum. Ob vires autem sollicitantes fieri potest, ut hae quatuor res immutentur, ideoque ad earum effectum cognoscendum necesse est, ut, quantum singulae tempusculo infinite parvo variantur, definire valeamus. Effectus ergo virium non tarn in his quatuor rebus, quam in earum variatione momentanea consistit, quam si assignare potuerimus, effectum perfecte cognoverimus; unde veritas Theorematis est manifesta.

#### COROLLARIUM 1

**786.** Quemadmodum ergo in motu punctorum effectus virium ex variatione celeritatis et directionis perfecte cognoscitur; ita in motu corporum rigidorum, praeter has binas variationes, ad centrum inertiae relatas, nosse oportet variationes, quas cum ipse axis gyrationis tum celeritas angularis subit.

#### COROLLARIUM 2

**787.** Sicut ergo vires definivimus, quibus motui gyratorio circa axem fixum data acceleratio inducatur, ita etiam vires definire licebit, quibus insuper ipse axis gyrationis datam variationem adipiscatur.

#### COROLLARIUM 3

**788.** Fundamentum ergo universae Theoriae de motu corporum rigidorum in hoc consistit, ut, quomodocunque vires sollicitantes fuerint comparatae, quaternas illas variationes temporis elemento productas assignare valeamus.

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 808

**SCHOLION 1**

**789.** Principia ad hunc finem ducentia in praecedentibus iam satis sunt exposita, ubi ostendimus, quomodo variationem tam in motu centri inertiae, quam in axe gyrationis eiusque motu determinari oportet. Verum quia hoc posterius opus, in quo summa huius Theoriae continetur, pluribus investigationibus innititur, quae saepe plurimum molestiae implicare solent, hic eas quasi in unum contrahens hanc Theoriam ita proponam, ut unico principio absolvi possit. Statim quidem hoc faciliiori modo uti potuissem, sicque non leves difficultates in superiori tractatione occurrentes evitavissem; verum in argumento adhuc tam parum tractato haud incongruum visum est methodum operosiorum et prolixiorum praemittere, quo singulae notiones in re pene nova animo firmiter imprimantur ipsaeque difficultates, quibus haec pars Mechanicae adhuc involuta videbatur, luculentius perspiciantur. Nihil vero minus hoc argumentum hic quasi de nova pertractabo neque ex hactenus allatis quicquam in subsidium vocabo.

**SCHOLION 2**

**790.** Cum igitur totum negotium huc reducatur, ut, quantae variationes in quaternis memoratis rebus a datis viribus producantur, definiatur; quoniam methodus directa hoc praestandi non patet, vice versa primum in vires inquiram, quae ad datas variationes momentaneas producendas sint necessariae, ut hinc vicissim ad id, quod quaerimus, reverti queamus. Et cum variatio in motu centri inertiae producta nihil habeat difficultatis, id tanquam in quiete spectabo; et cuiusmodi vires requirantur, investigabo, ut tam axis gyrationis, circa quem corpus iam gyrationis, quam celeritas angularis tempusculo infinite parvo datas variationes accipiant. Quoniam enim axis gyrationis cum celeritate angulari dari assumitur, motus singulorum elementorum corporis erit datus, qui si secundum ternas directiones fixas resolvatur, quantum hae ternae celeritates, tam ob variatam axis gyrationis positionem, quam ob celeritatis angularis variationem immutentur, colligere simulque vires hanc mutationem in singulis elementis corporis producentes assignare valebimus; atque his denique viribus elementaribus colligendis ipsas vires finitas quaesitas impetrabimus. Cum igitur primum motum singulorum corporis elementorum, dum corpus circa axem quemcunque per centrum inertiae transeuntem data celeritate angulari gyrationis, nosse debeamus, eius resolutionem secundum ternas directiones fixas, pro quibus ternos corporis axes principales assumam, in sequente problemate docebo.



**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

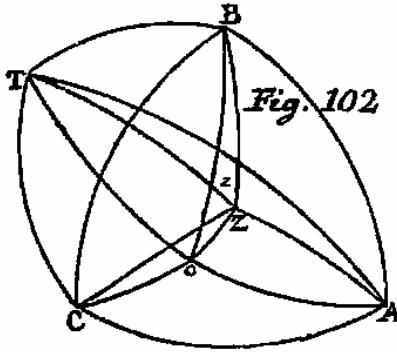
Translated and annotated by Ian Bruce.

page 809

**PROBLEMA 85**

**791.** Si corpus rigidum circa axem quemcunque per eius centrum inertiae transeuntem gyretur data celeritate, singulorum eius elementorum motum definire eumque secundum directiones axium principalium resolvere.

**SOLUTIO**



Circa centrum inertiae corporis  $I$ , quod in figura non est expressum, concipiatur descripta superficies sphaerica (Fig. 102), in qua sint  $A, B, C$  poli axium principalium, ita ut arcus  $AB, AC$  et  $BC$  sint quadrantes. Gyretur iam corpus circa axem quemcunque  $IO$  celeritate angulari  $= \gamma'$  in sensum  $ABC$  sintque pro gyrationis polo  $O$  arcus  $OA = \alpha, OB = \beta$  et  $OC = \gamma$ . Consideretur nunc corporis elementum quodcunque, a quo recta, ad centrum inertiae  $I$  ducta, superficiem sphaericam secet in  $Z$ ; eius autem distantia a centro  $I$  sit  $= r$ , dum radius sphaerae unitate exponitur; atque manifestum est motum eius elementi

similem fore motui puncti  $Z$ , dum nempe huius celeritas in ratione 1 ad  $r$  augetur. Quare sufficet motum puncti  $Z$  definivisse, pro quo si ad arcum  $OZ$  constituatur arcus  $ZzT$  normalis, erit  $Zz$  directio motus et celeritas  $= \gamma' \sin OZ$ , quoniam  $\sin OZ$  distantiam puncti  $Z$  ab axe gyrationis  $IO$  exprimit. Constituatur autem arcus  $ZT$  quadrans, ut radius  $IT$  fiat directioni motus  $Zz$  parallelus, ac iam celeritatem  $\gamma' \sin OZ$  secundum hanc directionem  $IT$  latam resolveri oportet secundum directiones axium principalium  $IA, IB, IC$ . Quem in finem ductis arcibus  $AT, BT, CT$ , qui illius rectae  $IT$  inclinationes ad hos axes metiuntur, obtinebitur

$$\text{celeritas secundum } IA = \gamma' \sin OZ \cdot \cos AT ,$$

$$\text{celeritas secundum } IB = \gamma' \sin OZ \cdot \cos BT$$

et

$$\text{celeritas secundum } IC = \gamma' \sin OZ \cdot \cos CT .$$

Iam quia arcus  $OT$  est pariter quadrans, ex triangulo  $AOT$  sit

$$\cos AT = \cos AOT \cdot \sin AO = -\sin AOZ \cdot \sin AO$$

ob  $TOZ = 90^\circ$ . Simili modo est

$$\cos BT = \cos BOT \cdot \sin BO = \sin BOZ \cdot \sin BO$$

$$\cos CT = \cos COT \cdot \sin CO = \sin COZ \cdot \sin CO .$$

At ob

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 810

$$\sin AZ : \sin AOZ = \sin OZ : \sin OAZ$$

erit

$$\sin AOZ \cdot \sin OZ = \sin AZ \cdot \sin OAZ$$

similique modo

$$\sin BOZ \cdot \sin OZ = \sin BZ \cdot \sin OBZ$$

et

$$\sin COZ \cdot \sin OZ = \sin CZ \cdot \sin OCZ ;$$

unde fit

$$\text{celeritas secundum } IA = -\gamma' \sin AO \cdot \sin AZ \cdot \sin OAZ$$

$$\text{celeritas secundum } IB = \gamma' \sin BO \cdot \sin BZ \cdot \sin OBZ$$

$$\text{celeritas secundum } IC = \gamma' \sin CO \cdot \sin CZ \cdot \sin OCZ .$$

Tum vero est

$$\sin BAO = \frac{\cos CO}{\sin AO}, \cos BAO = \frac{\cos BO}{\sin AO}$$

$$\sin BAZ = \frac{\cos CZ}{\sin AZ}, \cos BAZ = \frac{\cos BZ}{\sin AZ},$$

ergo

$$\sin OAZ = \frac{\cos CO \cdot \cos BZ - \cos BO \cdot \cos CZ}{\sin AO \cdot \sin AZ}$$

ideoque

$$\text{celerites secundum } IA = \gamma'(\cos BO \cdot \cos CZ - \cos CO \cdot \cos BZ) ;$$

similique modo reperitur

$$\text{celeritas secundum } IB = \gamma'(\cos CO \cdot \cos AZ - \cos AO \cdot \cos CZ) ,$$

$$\text{celeritas secundum } IC = \gamma'(\cos AO \cdot \cos BZ - \cos BO \cdot \cos AZ) ,$$

quae per  $r$  multiplicatae dabunt celeritates elementi propositi; pro quo si coordinatae axibus principalibus parallelae ponantur  $x, y, z$  erit

$$r \cos AZ = x, r \cos BZ = y \text{ et } r \cos OZ = z ;$$

quare ob

$$AO = \alpha, BO = \beta, CO = \gamma$$

erunt elementi propositi celeritates

$$\text{celeritas secundum } IA = \gamma'(z \cos \beta - y \cos \gamma)$$

$$\text{celeritas secundum } IB = \gamma'(x \cos \gamma - z \cos \alpha)$$

$$\text{celeritas secundum } IC = \gamma'(y \cos \alpha - x \cos \beta) .$$

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 811

**PROBLEMA 86**

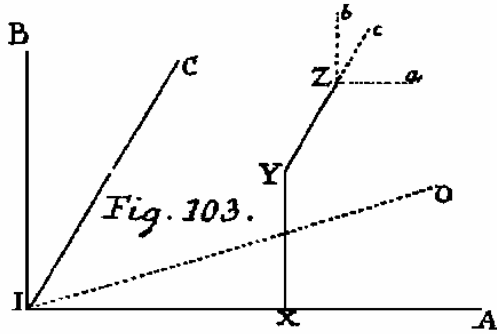
**792.** Si corpus rigidum gyretur circa axem quemcunque per eius centrum inertiae transeuntem data celeritate angulari, invenire vires elementares, quibus singula elementa sollicitari debent, ut elemento temporis  $dt$  tam ipse axis gyrationis quam celeritas angularis datas subeant variationes.

**SOLUTIO**

Sit  $I$  centrum inertiae corporis, eiusque axes principales  $IA, IB, IC$  gyreturque corpus circa axem quemcunque  $IO$  (Fig. 103), cuius ad quemlibet axem sit inclinatio

$$AIO = \alpha, BIO = \beta, CIO = \gamma,$$

celeritas autem angularis sit  $= \gamma'$  in sensum  $ABC$  directa; quae quantitates tempusculo  $dt$  crescere



debeant suis differentialibus  $d\alpha, d\beta, d\gamma$  et  $d\gamma'$ , ad quem effectum producendum vires elementares necessarias quaeri oporteat. Consideretur elementum corporis quodcunque  $dM$  in  $Z$  situm, pro quo sint coordinatae axibus principalibus parallelae

$$IX = x, XY = y, YZ = z,$$

vocenturque vires ad eius motum praescriptum efficiendum requisitae et secundum axes principales resolutae

$$Za = p, Zb = q \text{ et } Zc = r.$$

Secundum easdem directiones eius motus resolvatur, ponaturque celeritas secundum  $Za = u$ , secundum  $Zb = v$  et secundum  $Zc = w$ , atque cum ex primis motus principiis sit

$$du = \frac{2gpdt}{dM}, \quad dv = \frac{2gqdt}{dM}, \quad dw = \frac{2grdt}{dM},$$

vires quaesitae erunt:

$$p = \frac{dudM}{2gdt}, \quad q = \frac{dvdM}{2gdt}, \quad r = \frac{dwdM}{2gdt}.$$

Verum in praecedente problemate celeritates ternas  $u, v, w$  ita invenimus expressas, ut sit:

$$\begin{aligned} u &= \gamma'(z\cos\beta - y\cos\gamma) \\ v &= \gamma'(x\cos\gamma - z\cos\alpha) \\ w &= \gamma'(y\cos\alpha - x\cos\beta); \end{aligned}$$

quae quantum augeantur tempusculo  $dt$  cum ex variabilitate litterarum  $\gamma', d\alpha, d\beta, d\gamma$ , quae ut data spectatur, tum vero coordinatarum  $x, y, z$  iudicari oportet. Ad harum differentialia  $dx, dy, dz$  exhibent spatiola, per quae elementum  $dM$  tempusculo  $dt$  transferetur, ita ut sit

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 812

$$\begin{aligned} dx &= udt = \gamma'(z \cos \beta - y \cos \gamma) \\ dy &= vdt = \gamma'(x \cos \gamma - z \cos \alpha) \\ dz &= wdt = \gamma'(y \cos \alpha - x \cos \beta); \end{aligned}$$

Unde differentiatione rite instituta adipiscimur :

$$\begin{aligned} du &= d\gamma'(z \cos \beta - y \cos \gamma) - \gamma'(z d\beta \sin \beta - y d\gamma \sin \gamma) + \gamma' dt(w \cos \beta - v \cos \gamma) \\ dv &= \gamma'(x \cos \gamma - z \cos \alpha) - \gamma'(x d\gamma \sin \gamma - z d\alpha \sin \alpha) + \gamma' dt(u \cos \gamma - w \cos \alpha) \\ dw &= \gamma'(y \cos \alpha - x \cos \beta) - \gamma'(y d\alpha \sin \alpha - x d\beta \sin \beta) + \gamma' dt(v \cos \alpha - u \cos \beta). \end{aligned}$$

Cum vero sit

$$\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1$$

ideoque

$$\cos^2 \alpha + \cos^2 \beta = \sin^2 \gamma,$$

$$\cos^2 \alpha + \cos^2 \gamma = \sin^2 \beta$$

et

$$\cos^2 \beta + \cos^2 \gamma = \sin^2 \alpha,$$

hae formulae abeunt in istas:

$$\begin{aligned} du &= d\gamma'(z \cos \beta - y \cos \gamma) - \gamma' z d\beta \sin \beta + \gamma' y d\gamma \sin \gamma \\ &\quad + \gamma' \gamma' dt(y \cos \alpha \cos \beta + z \cos \gamma \cos \alpha - x \sin^2 \alpha) \\ dv &= \gamma'(x \cos \gamma - z \cos \alpha) - \gamma' x d\gamma \sin \gamma + \gamma' z d\alpha \sin \alpha \\ &\quad + \gamma' \gamma' dt(z \cos \beta \cos \gamma + x \cos \alpha \cos \beta - y \sin^2 \beta) \\ dw &= \gamma'(y \cos \alpha - x \cos \beta) - \gamma' y d\alpha \sin \alpha + \gamma' x d\beta \sin \beta \\ &\quad + \gamma' \gamma' dt(x \cos \gamma \cos \alpha + y \cos \beta \cos \gamma - y \sin^2 \gamma). \end{aligned}$$

ex quibus vires quaesitae elementares  $p, q, r$  innotescunt has scilicet formulas per

$\frac{dM}{2gdt}$  multiplicando.

**COROLLARIUM I**

**793.** Si igitur singula corporis elementa a talibus ternis viribus sollicitentur, dum corpus circa axem  $IO$  celeritate angulari  $\gamma'$  gyratur, elapso tempusculo  $dt$  celeritas angularis  $\gamma'$  augmentum accipiet  $= d\gamma'$  simulque axis gyrationis respectu axium principalium  $IA, IB, IC$  ita variabitur, ut anguli  $\alpha, \beta, \gamma$  suis differentialibus  $d\alpha, d\beta, d\gamma$  augeantur.

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 813

**COROLLARIUM 2**

**794.** Quatenus vires contemplamur idem corporis elementum  $dM$  sollicitantes, quantitates  $x, y, z$  in his formulis insunt tanquam constantes, quoniam iis situs elementi respectu axium principalium designatur, qui semper manet idem.

**COROLLARIUM 3**

**795.** Sin autem ab hoc elemento ad alia transire velimus, vires ea sollicitantes investigaturi, eadem quantitates  $x, y, z$  erunt variables et reliquae  $\alpha, \beta, \gamma, \gamma'$  cum suis differentialibus tanquam constantes spectandae, quoniam hae pro omnibus corporis elementis eodem instanti manent eadem.

**COROLLARIUM 4**

**796.** Quare si vires omnia elementa sollicitantes in unam summam colligere, velimus, hae tantum formulae  $\int x dM, \int y dM$  et  $\int z dM$  integrandae occurrunt; quarum differentialia cum evanescant, ob  $I$  centrum inertiae corporis, patet summas omnium virium  $p$ , item  $q$  et  $r$  seorsim evanescere.

**SCHOLION**

**797.** Quia summae omnium virium  $p, q$  et  $r$  evanescent, quod semper evenire debet, quamdiu centrum inertiae in quiete persistit, earum effectua tantum ex earum momentis est diiudicandus; atque aliae quaeque vires eadem momenta habentes eundem effectum producent, dummodo illis aequales et contrariae in centro inertiae applicentur. Verum hic non sufficit, ut vires idem habeant momentum respectu unius cuiuspiam axis, sed necesse est, ut respectu omnium plane axium eadem momenta producant, alioquin non pro aequivalentibus essent habendae. Hoc autem evenit, dummodo pro tribus axibus principalibus eadem momenta suppeditent, id quod sequente propositione extra dubium collocabitur.

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 814

**PROBLEMA 87**

**798.** Dum corpus rigidum circa axem quemcunque per centrum inertiae transeuntem data celeritate angulari gyratur, definire virium momenta respectu trium axium principalium, quibus tam ipsi axi gyrationis quam celeritati angulari data immutatio inducatur.

**SOLUTIO**

Manentibus pro axe gyrationis  $IO$  angulis

$$AIO = \alpha, BIO = \beta, CIO = \gamma,$$

(Fig. 103) circa quem corpus iam gyretur celeritate angulari  $\gamma'$  in sensum  $ABC$ , haeque quantitates tempusculo  $dt$  differentialibus suis crescere debeant, considerentur pro elemento corporis quocunque  $dM$  in  $Z$  coordinatis

$$IX = x, XY = y \text{ et } YZ = z$$

determinato vires elementares ante definitae

$$Za = p = \frac{du \, dM}{2gdt}, \quad Zb = q = \frac{dv \, dM}{2gdt}, \quad Zc = r = \frac{dw \, dM}{2gdt};$$

ex quibus respectu axis  $IA$  oritur momentum in sensum  $BC$

$$= ry - qz = \frac{dM}{2gdt} (ydw - zdv),$$

at respectu axis  $IB$  momentum in sensum  $CA$

$$= pz - rx = \frac{dM}{2gdt} (zdu - xdw),$$

ac denique respectu axis  $IC$  momentum in sensum  $AB$

$$= qx - py = \frac{dM}{2gdt} (xdv - ydu).$$

Quodsi hic pro  $du, dv, dw$  formulas ante inventas substituamus, reperiemus:

$$ydw - zdv = d\gamma' \left( (y \, y + zz) \cos \alpha - xy \cos \beta - xz \cos \gamma \right)$$

$$- \gamma' (yy + zz) d\alpha \sin \alpha + \gamma' xy d\beta \sin \beta + \gamma' xz d\gamma \sin \gamma$$

$$+ \gamma' \gamma' dt \left( (yy - zz) \cos \beta \cos \gamma + xy \cos \alpha \cos \gamma - xz \cos \alpha \cos \beta - yz (\sin^2 \gamma - \sin^2 \beta) \right)$$

$$zdu - xdw = d\gamma' \left( (xx + zz) \cos \beta - yz \cos \gamma - xy \cos \alpha \right)$$

$$- \gamma' (xx + zz) d\beta \sin \beta + \gamma' yz d\gamma \sin \gamma + \gamma' xy d\alpha \sin \alpha$$

$$+ \gamma' \gamma' dt \left( (zz - xx) \cos \alpha \cos \gamma + yz \cos \alpha \cos \beta - xy \cos \beta \cos \gamma - xz (\sin^2 \alpha - \sin^2 \gamma) \right)$$

$$xdv - ydu = d\gamma' \left( (xx + yy) \cos \gamma - xz \cos \alpha - yz \cos \beta \right)$$

$$- \gamma' (xx + yy) d\gamma \sin \gamma + \gamma' xz d\alpha \sin \alpha + \gamma' yz d\beta \sin \beta$$

$$+ \gamma' \gamma' dt \left( (xx - yy) \cos \alpha \cos \beta + xz \cos \beta \cos \gamma - yz \cos \alpha \cos \gamma - xy (\sin^2 \beta - \sin^2 \alpha) \right).$$

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 815

Multiplicentur iam hae formulae per  $\frac{dM}{2gdt}$  et per totam corporis molem integrentur; quem in finem sint  $Maa, Mbb, Mcc$  momenta inertiae respectu axium principalium  $IA, IB, IC$  et, cum sit

$$\begin{aligned}\int xxdM &= \frac{1}{2}M(bb+cc-aa), & \int yzdM &= 0, \\ \int yydM &= \frac{1}{2}M(aa+cc-bb), & \int xzdM &= 0, \\ \int zzdM &= \frac{1}{2}M(aa+bb-cc), & \int xydM &= 0,\end{aligned}$$

obtinebimus terna virium momenta respectu axium principalium, quibus effectua praescriptus producitur, ita expressa:

I. Momentum virium respectu axis  $IA$  in sensum  $BC$

$$\frac{dM}{2gdt}(aad\gamma' \cos \alpha - \gamma' aad\alpha \sin \alpha + \gamma' \gamma'(cc-bb) dt \cos \beta \cos \gamma),$$

II. Momentum virium respectu axis  $IB$  in sensum  $CA$

$$\frac{dM}{2gdt}(bbd\gamma' \cos \beta - \gamma' bbd\beta \sin \beta + \gamma' \gamma'(aa-cc) dt \cos \alpha \cos \gamma),$$

III. Momentum virium respectu axis  $IC$  in sensum  $AB$

$$\frac{dM}{2gdt}(ccd\gamma' \cos \gamma - \gamma' ccd\gamma \sin \gamma + \gamma' \gamma'(bb-aa) dt \cos \alpha \cos \beta).$$

**COROLLARIUM I**

**799.** Ut ergo corpus circa eundem axem uniformiter gyretur, terna momenta virium ob  
 $d\gamma' = 0, d\alpha = 0, d\beta = 0, d\gamma = 0$   
erunt

$$\begin{aligned}\text{I.} &= \frac{M\gamma'\gamma'(cc-bb)\cos\beta\cos\gamma}{2g}, \\ \text{II.} &= \frac{M\gamma'\gamma'(aa-cc)\cos\alpha\cos\gamma}{2g}, \text{ et} \\ \text{III.} &= \frac{M\gamma'\gamma'(bb-aa)\cos\alpha\cos\beta}{2g},\end{aligned}$$

quae, nisi axis gyrationis in aliquem axium principalium incidat, non evanescent.

**COROLLARIUM 2**

**800.** Simili modo intelligitur, quibusnam viribus sit opus, ut vel sola celeritas angularis mutetur vel sola axis gyrationis positio varietur; scilicet vires, quarum momenta cum ante definitis conveniant, hoc praestabunt, si modo illis aequales et contrariae in centro inertiae applicentur, ut ipsae vires pro evanescentibus haberi queant totusque effectus solis earum momentis debeat.

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 816

**COROLLARIUM 3**

**801.** Si corpus circa ipsum axem principalem  $IA$  celeritate angulari  $\gamma'$  gyretur, quae suo differentiali  $d\gamma'$  augeri debeat, ob  $\alpha = 0$  et  $\beta = \gamma = 90^\circ$  ad hoc tantum respectu axis  $IA$  requiritur momentum virium  $= \frac{dMaad\gamma'}{2gdt}$ , uti iam supra invenimus.

**SCHOLION**

**802.** Problema hoc haud difficilius solutu fuisset, si corpori praeter motum gyrationum insuper motum progressivum quemcunque tribuissemus, qui tempusculo  $dt$  etiam praescripto modo variari deberet; si enim centrum inertiae motum habeat quemcunque, qui secundum axes principales resolutus praebet celeritates  $l, m, n$ , tempusculo  $dt$  suis quoque differentialibus augendas, celeritates  $u, v, w$  supra valores ex motu gyrationis natos his progressivis  $l, m, n$  augeri deberent, atque ex harum incrementis nascerentur vires, quarum aequivalens per centrum inertiae transiret pariterque se haberet, ac si corpus sine ullo motu gyrationis hunc solum motum progressivum prosequi deberet. Quo id confirmatur, quod iam supra ostendimus, in tali motu mixto semper motum progressivum et gyrationis separari et utrumque seorsim, quasi alter non adesset, considerari ac determinari licere.

**PROBLEMA 88**

**803.** Si corpus rigidum, dum circa datum axem  $IO$  data celeritate angulari  $= \gamma'$  gyratur, a viribus quibuscunque sollicitetur, quibus simul aequales et contrariae ipsi centro inertiae sint applicatae, determinare tam variationem axis, quam mutationem celeritatis angularis elemento temporis  $dt$  productum.

**SOLUTIO**

Colligantur virium sollicitantium momenta respectu ternorum axium principalium corporis sitque

$$\begin{aligned} \text{momentum virium respectu axis } IA \text{ in sensum } BC &= P, \\ \text{momentum virium respectu axis } IB \text{ in sensum } CA &= Q, \\ \text{momentum virium respectu axis } IC \text{ in sensum } AB &= R. \end{aligned}$$

Momenta autem inertiae corporis respectu eorundem axium sint ut hactenus  $Maa, Mbb, Mcc$ .

Quodsi iam corpus gyretur in sensum  $ABC$  celeritate angulari  $= \gamma'$  circa axem  $IO$ , cuius inclinationes ad eosdem axes principales nunc sint

$$AIO = \alpha, BIO = \beta, CIO = \gamma,$$

hae quantitates tempusculo  $dt$  sequentes mutationes subibunt:

$$\begin{aligned} \frac{2gPdt}{Maa} &= d\gamma' \cos \alpha - \gamma' d\alpha \sin \alpha + \frac{cc - bb}{aa} \gamma' \gamma' dt \cos \beta \cos \gamma \\ \frac{2gPdt}{Mbb} &= d\gamma' \cos \beta - \gamma' d\beta \sin \beta + \frac{aa - cc}{bb} \gamma' \gamma' dt \cos \alpha \cos \gamma \\ \frac{2gPdt}{Mcc} &= d\gamma' \cos \gamma - \gamma' d\gamma \sin \gamma + \frac{bb - aa}{cc} \gamma' \gamma' dt \cos \alpha \cos \beta, \end{aligned}$$



**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 817

ex quibus aequationibus quaternae incognitae  $\alpha, \beta, \gamma$ , et  $\gamma'$  determinantur, quoniam tantum pro tribus sunt habendae ob

$$\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1 .$$

Cum igitur sit

$$d\alpha \sin \alpha \cos \alpha + d\beta \sin \beta \cos \beta + d\gamma \sin \gamma \cos \gamma = 0 , ,$$

si prima per  $\cos \alpha$ , secunda per  $\cos \beta$ , tertia per  $\cos \gamma$  multiplicetur, productis addendis prodibit:

$$\begin{aligned} d\gamma' + \left( \frac{cc-bb}{aa} + \frac{aa-cc}{bb} + \frac{bb-aa}{cc} \right) \gamma' \gamma' dt \cos \alpha \cos \beta \cos \gamma \\ = \frac{2gdt}{M} \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right) \end{aligned}$$

seu

$$\begin{aligned} d\gamma' = \frac{(cc-bb)(aa-cc)(bb-aa)}{aabbcc} \gamma' \gamma' dt \cos \alpha \cos \beta \cos \gamma \\ + \frac{2gdt}{M} \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right), \end{aligned}$$

quo valore substituto obtinebuntur hae aequationes:

$$\begin{aligned} \gamma' d\alpha \sin \alpha &= \frac{(cc-bb)}{aa} \gamma' \gamma' dt \cos \beta \cos \gamma \left( 1 + \frac{(aa-cc)(bb-aa)}{bbcc} \cos^2 \alpha \right) \\ &\quad + \frac{2gdt}{M} \left( \frac{Q \cos \alpha \cos \beta}{bb} + \frac{R \cos \alpha \cos \gamma}{cc} - \frac{P \sin^2 \alpha}{aa} \right) \\ \gamma' d\beta \sin \beta &= \frac{(aa-cc)}{bb} \gamma' \gamma' dt \cos \alpha \cos \gamma \left( 1 + \frac{(bb-aa)(cc-bb)}{aacc} \cos^2 \beta \right) \\ &\quad + \frac{2gdt}{M} \left( \frac{R \cos \beta \cos \gamma}{cc} + \frac{P \cos \beta \cos \alpha}{aa} - \frac{Q \sin^2 \beta}{bb} \right) \\ \gamma' d\gamma \sin \gamma &= \frac{(bb-aa)}{cc} \gamma' \gamma' dt \cos \alpha \cos \beta \left( 1 + \frac{(cc-bb)(aa-cc)}{aabb} \cos^2 \gamma \right) \\ &\quad + \frac{2gdt}{M} \left( \frac{P \cos \gamma \cos \alpha}{aa} + \frac{Q \cos \gamma \cos \beta}{bb} - \frac{R \sin^2 \gamma}{cc} \right). \end{aligned}$$

At si prima illarum aequationum per  $aa \cos \alpha$ . secunda per  $bb \cos \beta$ , tertia per  $cc \cos \gamma$  multiplicetur, eas addendo orietur

$$\begin{aligned} \frac{2gdt}{M} (P \cos \alpha + Q \cos \beta + R \cos \gamma) = d\gamma' (aa \cos^2 \alpha + bb \cos^2 \beta + \cos^2 \gamma) \\ - \gamma' (aada \sin \alpha \cos \alpha + bbd\beta \sin \beta \cos \beta + ccd\gamma \sin \gamma \cos \gamma), \end{aligned}$$

quae per  $2M \gamma'$  multiplicata et ex altera parte integrata dat

## EULER'S

### *Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.*

#### Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 818

$$M \gamma' \gamma' (aa \cos^2 \alpha + bb \cos^2 \beta + cc \cos^2 \gamma) = 4g \int \gamma' dt (P \cos \alpha + Q \cos \beta + R \cos \gamma),$$

quae quantitas exprimit corporis vim vivam.

#### COROLLARIUM 1

**804.** Si igitur, dum circa axem quemeunque per centrum inertiae transeuntem gyatur, a viribus quibuscunque sollicitetur, hinc variationes momentaneae tam in situ axis gyrationis respectu axium principalium, quam in celeritate angulari determinantur.

#### COROLLARIUM 2

S05. Si corpus a nullis plane viribus externis sollicitetur, axis gyrationis cum celeritate angulari ita variantur, ut sit:

$$\begin{aligned} \text{I. } d\gamma' &= \frac{(cc-bb)(aa-cc)(bb-aa)}{aabbcc} \gamma' \gamma' dt \cos \alpha \cos \beta \cos \gamma \\ \text{II. } d\alpha \sin \alpha &= \frac{cc-bb}{aa} \gamma' dt \cos \beta \cos \gamma \left( 1 + \frac{(aa-cc)(bb-aa)}{bbcc} \cos^2 \alpha \right) \\ \text{III. } d\beta \sin \beta &= \frac{aa-cc}{bb} \gamma' dt \cos \alpha \cos \gamma \left( 1 + \frac{(bb-aa)(cc-bb)}{aacc} \cos^2 \beta \right) \\ \text{IV. } d\gamma \sin \gamma &= \frac{bb-aa}{cc} \gamma' dt \cos \alpha \cos \beta \left( 1 + \frac{(cc-bb)(aa-cc)}{aabb} \cos^2 \gamma \right) \end{aligned}$$

et vis viva  $M \gamma' \gamma' (aa \cos^2 \alpha + bb \cos^2 \beta + cc \cos^2 \gamma)$  perpetuo manet constans.

#### COROLLARIUM 3

**806.** Si corpus quiescat, ut sit  $\gamma' = 0$ , ex momentis virium  $P, Q, R$  respectu axium principalium sumtis, axis, circa quem corpus primum gyrationis incipiet, ex his aequationibus definietur:

$$\frac{Q \cos \alpha \cos \beta}{bb} + \frac{R \cos \alpha \cos \gamma}{cc} - \frac{P \sin^2 \alpha}{aa} = 0$$

seu

$$\begin{aligned} \frac{P}{aa} &= \cos \alpha \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right), \\ \frac{R \cos \beta \cos \gamma}{cc} + \frac{P \cos \beta \cos \alpha}{aa} - \frac{Q \sin^2 \beta}{bb} &= 0 \end{aligned}$$

seu

$$\begin{aligned} \frac{Q}{bb} &= \cos \beta \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right), \\ \frac{P \cos \gamma \cos \alpha}{aa} + \frac{Q \cos \gamma \cos \beta}{bb} - \frac{R \sin^2 \gamma}{cc} &= 0 \end{aligned}$$

seu

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 819

$$\frac{R}{cc} = \cos \gamma \left( \frac{P \cos \alpha}{aa} + \frac{Q \cos \beta}{bb} + \frac{R \cos \gamma}{cc} \right),$$

unde, cum sit

$$\cos \alpha : \cos \beta : \cos \gamma = \frac{P}{aa} : \frac{Q}{bb} : \frac{R}{cc},$$

erit

$$\cos \alpha = \frac{P}{aa} \cdot \sqrt{\left( \frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4} \right)}$$

$$\cos \beta = \frac{Q}{bb} \cdot \sqrt{\left( \frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4} \right)}$$

$$\cos \gamma = \frac{R}{cc} \cdot \sqrt{\left( \frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4} \right)},$$

ac tempusculo  $dt$  fiet

$$d\gamma' = \frac{2gdt}{M} \sqrt{\left( \frac{PP}{a^4} + \frac{QQ}{b^4} + \frac{RR}{c^4} \right)}.$$

**SCHOLION 1**

**807.** In hoc ergo solo problemate omnia continentur, quae supra [§ 639] per multas ambages magno labore elicuimus, cum tamen hic nonnisi primis motus principiis simus usi omniaque sint maxime perspicua. Ita eum supra, dum corpus quiescit, axem, circa quem ipsi vires primum motum gyratorium imprimunt, vehementer operose determinavisse, hic ista determinatio instar corollarii ex praesente problemate sponte fluxit; cuius consensus cum superiori quo facilius perspiciatur, ac ne ambiguitas signi radicalis moram facessat, sit iterum pro axe gyrationis  $IF$  (Fig. 82, voluminis praecedentis) angulus  $AIE = \eta$  et angulus  $EIF = \vartheta$ , erit

$$\cos \alpha = \cos \eta \cos \vartheta,$$

$$\cos \beta = -\sin \eta \cos \vartheta$$

et

$$\cos \gamma = \sin \vartheta,$$

unde ob

$$\text{tang } \eta = \frac{-\cos \beta}{\cos \alpha}$$

erit

$$\text{tang } \eta = \frac{-Qaa}{Pbb}$$

et

$$\text{tang } \vartheta = \frac{\cos \gamma}{\cos \alpha} \cos \eta = \frac{Raa}{Pcc} \cos \eta.$$

Cum autem vires sollicitantes ibi sint

$$VP = P, VQ = Q, VR = R$$

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 820

existente angulo  $AIV = \delta$  et  $IV = h$ , erit harum virium momentum respectu axis  $IA$  in sensum  $BC = Rh \sin \delta$ , quod hic nobis est  $P$ ; tum earum momentum respectu axis  $IB$  in sensum  $CA = -Rh \cos \delta$ , quod hic nobis est  $Q$ , et momentum respectu axis  $IC$  in sensum  $AB = Qh \cos \delta - Ph \sin \delta$ ,

quod hic nobis est  $R$ . Quibus valoribus pro  $P$ ,  $Q$  et  $R$  positis habebimus prorsus ut supra

$$\text{tang } \eta = \frac{aa \cos \delta}{bb \sin \delta}$$

et

$$\text{tang } \vartheta = \frac{aa(Q \cos \delta - P \sin \delta)}{ccR \sin \delta} \cos \eta.$$

Deinde etiam, quae supra de variatione momentanea motus gyratorii, dum corpus a nullis viribus sollicitatum circa axem non principalem gyrat, per nimis intricata ratiocinia tandem eruimus, hic positis virium momentis  $P = 0$ ,  $Q = 0$ ,  $R = 0$  fiunt planissima, uti in corollario 2 ostendimus. Quae autem supra vix attingere ausi fueramus, cum corpus insuper a viribus quibuscunque sollicitatur, hic pari facilitate eodemque labore feliciter expeditimus, ita ut in hoc tantum capite a primis motus principiis profecti universam Theoriam motus corporum rigidorum perfecte condidisse videamur.

**SCHOLION 2**

**808.** Cum autem propositis viribus sollicitantibus quibuscunque, quarum momenta respectu axium principalium in sensum  $ABC$  sumta sint  $P$ ,  $Q$ ,  $R$ , totum negotium in determinatione ternorum angulorum  $\alpha$ ,  $\beta$ ,  $\gamma$  et celeritatis  $\gamma'$  versetur, pro quo ternas invenimus aequationes, quandoquidem anguli illi relationem inter se tenent, aequationes illae levi substitutione multo commodiores reddi possunt. Quodsi enim, quia litteris  $x$ ,  $y$ ,  $z$  ad indolem corporis indicandam non amplius indigemus, ponamus

$$\gamma' \cos \alpha = x, \quad \gamma' \cos \beta = y, \quad \gamma' \cos \gamma = z,$$

omnes anguli ex calculo elidentur summaque totius Theoriae motus corporum rigidorum his tribus formulis satis simplicibus continebitur:

$$\begin{aligned} dx + \frac{cc-bb}{aa} yz dt &= \frac{2gPdt}{Maa} \\ dy + \frac{aa-cc}{bb} zx dt &= \frac{2gQdt}{Mbb} \\ dz + \frac{bb-aa}{cc} xy dt &= \frac{2gRdt}{Mcc}. \end{aligned}$$

Quare si corpus a nullis viribus sollicitetur, statim colligimus  $aaxdx + bbydy + cczdz = 0$

seu

$$aaxx + bbyy + cczz = \text{Const.}$$

Tum vero ex binis  $dt$  elidendo erit

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 821

$$\frac{aadx}{bbdy} = \frac{(cc - bb)y}{(aa - cc)x}$$

ideoque integrando

$$\frac{aa}{cc - bb} xx = \frac{bb}{aa - cc} yy + \text{Const.}$$

Quare si initio fuerit

$$x = \mathfrak{A}, \quad y = \mathfrak{B}, \quad z = \mathfrak{C}$$

ponamusque

$$\frac{aa}{cc - bb} = A, \quad \frac{bb}{aa - cc} = B \quad \text{et} \quad \frac{cc}{bb - aa} = C,$$

habebitur

$$Axx - Byy = A\mathfrak{A}^2 - B\mathfrak{B}^2$$

et

$$Axx - Czz = A\mathfrak{A}^2 - C\mathfrak{C}^2$$

ideoque

$$y = \frac{\sqrt{(Axx - A\mathfrak{A}^2 + B\mathfrak{B}^2)}}{\sqrt{B}}$$

et

$$z = \frac{\sqrt{(Axx - A\mathfrak{A}^2 + C\mathfrak{C}^2)}}{\sqrt{C}}.$$

Quare cum sit

$$Adx + yzdt = 0,$$

fiet

$$dt = - \frac{Adx\sqrt{BC}}{\sqrt{(Axx - A\mathfrak{A}^2 + B\mathfrak{B}^2)}\sqrt{(Axx - A\mathfrak{A}^2 + C\mathfrak{C}^2)}}$$

sicque etiam hoc problema, quod supra non parum molestiae creaverat, satis expedite est solutum.

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 822

**PROBLEMA 89**

**809.** Si ad quodvis tempus noverimus axem gyrationis respectu axium principalium una cum celeritate angulari corporis circa hunc axem, definire ad quodvis tempus situm axium principalium respectu spatii absoluti.

**SOLUTIO**

In spatio absoluto concipiatur sphaera immobilis, in cuius centro versetur corporis centrum inertiae  $I$  (Fig. 89) in eaque assumatur circulus fixus maximus  $ZXVY$  in eoque punctum fixum  $Z$ , quo situs axium principalium quovis tempore referatur. Ac nunc quidem elapso tempore  $t$  respondeant corporis axes principales in sphaera immobili punctis  $A, B, C$ , a quibus si ad  $Z$  ducantur arcus circulorum maximorum, vocentur ii

$$ZA = l, \quad ZB = m \quad \text{et} \quad ZC = n,$$

tum vero sint anguli

$$XZA = \lambda, \quad XZB = \mu \quad \text{et} \quad XZC = \nu.$$

Nunc autem reperiatur axis gyrationis in  $O$ , ut sit

$$AO = \alpha, \quad BO = \beta \quad \text{et} \quad CO = \gamma,$$

circa quem corpus gyretur in sensum  $ABC$  celeritate angulari  $\gamma'$ ; tempusculo ergo  $dt$  polus  $A$  vertetur per arculum

$$Aa = \gamma' dt \sin \alpha$$

existente  $Aa$  ad arcum  $OA$  normali, ita ut sit

$$\sin BAa = \frac{\cos \beta}{\sin \alpha} \quad \text{et} \quad \cos BAa = \frac{\cos \gamma}{\sin \alpha}.$$

At est

$$\sin ZAB = -\frac{\cos n}{\sin l} \quad \text{et} \quad \cos ZAB = \frac{\cos m}{\sin l},$$

unde colligitur

$$\sin ZAa = \frac{\cos \beta \cos m + \cos \gamma \cos n}{\sin \alpha \sin l},$$

$$\cos ZAa = \frac{\cos \gamma \cos m - \cos \beta \cos n}{\sin \alpha \sin l}.$$

Ducto iam ex  $a$  ad arcum  $ZA$  perpendiculo  $a\alpha$ , erit

$$A\alpha = \frac{\gamma' dt}{\sin l} (\cos \gamma \cos m - \cos \beta \cos n)$$

et

$$a\alpha = \frac{\gamma' dt}{\sin l} (\cos \beta \cos m + \cos \gamma \cos n).$$

Verum est

$$A\alpha = -dl \quad \text{et} \quad a\alpha = -d\lambda \sin l$$

ideoque hinc et ob analogiam sequentes concluduntur differentialium valores:

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 823

$$\begin{aligned} dl \sin l &= \gamma' dt(\cos \beta \cos n - \cos \gamma \cos m) \\ d\lambda \sin^2 l &= -\gamma' dt(\cos \beta \cos m + \cos \gamma \cos n) \\ dm \sin m &= \gamma' dt(\cos \gamma \cos l - \cos \alpha \cos n) \\ d\mu \sin^2 m &= -\gamma' dt(\cos \gamma \cos n + \cos \alpha \cos l) \\ dn \sin n &= \gamma' dt(\cos \alpha \cos m - \cos \beta \cos l) \\ dv \sin^2 n &= -\gamma' dt(\cos \alpha \cos l + \cos \alpha \cos m). \end{aligned}$$

Harum autem ternarum priorum binas resolvisse sufficit, cum sit

$$\cos^2 l + \cos^2 m + \cos^2 n = 1,$$

iisque resolutis unica reliquarum totum negotium absolvit.

**COROLLARIUM 1**

**810.** Si ponamus

$$\gamma' \cos \alpha = x, \quad \gamma' \cos \beta = y \quad \text{et} \quad \gamma' \cos \gamma = z,$$

ut sit ex momentis virium sollicitantium  $P, Q, R$ :

$$\begin{aligned} dx + \frac{cc - bb}{aa} yz dt &= \frac{2gPdt}{Maa}, \\ dy + \frac{aa - cc}{bb} xz dt &= \frac{2gQdt}{Mbb}, \\ dz + \frac{bb - aa}{cc} xy dt &= \frac{2gRdt}{Mcc}, \end{aligned}$$

nunc sequentes aequationes adiungi oportet:

$$\begin{aligned} dl \sin l &= dt(y \cos n - z \sin m) \\ dm \sin m &= dt(z \cos l - x \cos n) \\ dn \sin n &= dt(x \cos m - y \cos l) \\ d\lambda \sin^2 l &= -dt(y \cos m + z \cos n) \\ d\mu \sin^2 m &= -dt(z \cos n + x \cos l) \\ dv \sin^2 n &= -dt(x \cos l + y \cos m). \end{aligned}$$

**COROLLARIUM 2**

**811.** Si porro ponamus

$$\cos l = p, \quad \cos m = q, \quad \cos n = r,$$

posteriores aequationes has induent formas ob  $pp + qq + rr = 1$ :

$$\begin{aligned} dp + dt(yr - zq) &= 0 \\ dq + dt(zp - xr) &= 0 \\ dr + dt(xq - yp) &= 0 \end{aligned}$$

**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 824

$$d\lambda + \frac{dt(yq+ zr)}{qq+ rr} = 0$$

$$d\mu + \frac{dt(zr+ xp)}{pp+ rr} = 0$$

$$dv + \frac{dt(xp+ yq)}{pp+ qq} = 0,$$

unde etiam fit

$$xdp + ydq + zdr = 0 ,$$

quemadmodum est

$$pdp + qdq + rdr = 0 .$$

**SCHOLION**

**812.** Etsi hic problema praecedens, quasi iam esset solutum, spectavi, tamen plerumque ambo problemata coniungi eorumque resolutionem simul institui oportet, quemadmodum in praecedente capite de motu turbinum usu venit. Haec scilicet amborum problematum coniunctio necessaria est, quando vires sollicitantes a situ corporis absoluto pendent, quod quidem, si vires externae affuerint, semper contingere solet. His igitur casibus momenta virium  $P, Q, R$  arcus  $l, m, n$  ac fortasse etiam angulos  $\lambda, \mu, \nu$  involvent, ita ut omnes aequationes corollario 1 exhibitae simul perpendi debeant, antequam solutio suscipi queat. Quodsi corpus insuper motu progressivo feratur, fieri solet, ut vires etiam ab eo pendeant, ex quo formulas motum progressivum involventes simul ad reliquas adiici oportebit, quibus casibus solutio maxime complicata reddetur. Nunc igitur his problematibus expeditis problema generale de motu libero corporum rigidorum a viribus quibuscunque sollicitatorum aggredi poterimus.

**PROBLEMA 90**

**813.** Si corpus rigidum initio quomodocunque proiectum deinceps a viribus quibuscunque sollicitetur, quarum actioni libere obsequi queat, eius motum determinare.

**SOLUTIO**

Quod primo ad eius motum progressivum seu motum, quo centrum inertiae promovetur, attinet, is per eadem praecepta, quae pro motu punctorum sunt tradita, definietur. Scilicet tota corporis massa, quae sit =  $M$ , in eius centro inertiae collecta concipiatur, ac singulis momentis omnes vires, quibus corpus sollicitatur, secundum suam quaeque directionem ipsi centro inertiae applicentur; ut habeatur casus puncti, cuius autem massa finita est censenda =  $M$ , a viribus sollicitati, cuius propterea motus per praecepta supra tradita determinari vel saltem formulis analyticis exprimi poterit, nulla habita ratione motus gyratorii, quo interea forte corpus circa centrum inertiae agitetur. Tum vero ad hunc motum investigandum, priori motu progressivo penitus seposito, centrum inertiae iam ut quiescens consideretur; ac primo quidem corporis terni axes principales explorentur, qui ex centro inertiae  $I$  educti sint  $IA, IB, IC$ , eorumque respectu momenta inertiae  $Maa, Mbb, Mcc$ ; quibus



**EULER'S**  
**Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.**  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 825

cognitis sphaera concipiatur immobilis circa centrum inertiae  $I$  descripta, in qua tam circulus maximus  $ZXVY$  quam in eo punctum  $Z$  fixum assumatur, quo situs corporis quovis tempore referatur. Nunc igitur elapso tempore  $= t$  teneat corpus ob motum gyratorium situm in figura repraesentatum, in quo axes principales respondeant in superficie sphaerica punctis  $A, B, C$  quadrantis intervallo a se invicem distantibus, pro quorum situ praesente ponatur

$$\text{arcus } ZA = l, \quad ZB = m \text{ et } ZC = n$$

item

$$\text{anguli } XZA = \lambda, \quad XZB = \mu, \quad XZC = \nu,$$

qui quomodo a se invicem pendeant, ex sphaericis est manifestum. Porro gyretur nunc corpus circa axem  $IO$  celeritate angulari  $\gamma'$  in sensum  $ABC$ , ac pro situ huius axis ponantur arcus

$$AD = \alpha, \quad BD = \beta, \quad CO = \gamma;$$

atque hae sunt quantitates per sua differentialia ita determinanda, ut posito  $t = 0$  statui corporis initiali convenient. Ad hoc considerentur vires corpus nunc sollicitantes, quarum colligantur momenta inertiae respectu axium principalium corporis, sitque

$$\text{momentum virium respectu axis } IA \text{ in sensum } BC = P,$$

$$\text{momentum virium respectu axis } IB \text{ in sensum } CA = Q,$$

$$\text{momentum virium respectu axis } IC \text{ in sensum } AB = R,$$

atque ponendo brevitatis gratia

$$\gamma' \cos \alpha = x, \quad \gamma' \cos \beta = y \quad \text{et} \quad \gamma' \cos \gamma = z,$$

ut sit

$$\gamma' = \sqrt{(x^2 + y^2 + z^2)},$$

supra invenimus fore:

$$dx + \frac{cc - bb}{aa} yz dt = \frac{2gPdt}{Maa},$$

$$dy + \frac{aa - cc}{bb} xz dt = \frac{2gQdt}{Mbb},$$

$$dz + \frac{bb - aa}{cc} xy dt = \frac{2gRdt}{Mcc},$$

quibuscum posito

$$\cos l = p, \quad \cos m = q \quad \text{et} \quad \cos n = r$$

coniungantur hae aequationes:

**EULER'S**  
*Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.*  
 Chapter FIVETEEN.

Translated and annotated by Ian Bruce.

page 826

$$dp + dt(yr - zq) = 0$$

$$dq + dt(zp - xr) = 0$$

$$dr + dt(xq - yp) = 0$$

$$d\lambda + \frac{dt(yq + zr)}{qq + rr} = 0$$

$$d\mu + \frac{dt(zr + xp)}{rr + pp} = 0$$

$$dv + \frac{dt(xp + yq)}{pp + qq} = 0,$$

quae si ita resolvi et integrari queant, ut ad quodvis tempus  $t$  assignari possint quantitates  $x, y, z, p, q, r, \lambda, \mu, \nu$ , problema erit perfecte solutum. In his postremis autem aequationibus notandum est esse

$$pp + qq + rr = 1,$$

unde

$$pdp + qdq + rdr = 0,$$

tum vero etiam

$$xdp + ydq + zdr = 0.$$

Denique

$$\sin(\mu - \lambda) = -\frac{\cos n}{\sin l \sin m}$$

et

$$\cos(\mu - \lambda) = -\frac{\cos l \cos m}{\sin l \sin m},$$

ideoque

$$\text{tang}(\mu - \lambda) = \frac{r}{pq}$$

et

$$\text{tang}(\nu - \mu) = \frac{p}{qr},$$

tum

$$\text{tang}(\lambda - \nu) = \frac{q}{pr}$$

seu

$$\text{tang}(\nu - \lambda) = -\frac{q}{pr},$$

ita ut sufficiat angulorum  $\lambda, \mu, \nu$  unicum invenisse.

**EULER'S**  
***Theoria Motus Corporum Solidorum Seu Rigidorum VOL. 2.***  
*Chapter FIVETEEN.*

Translated and annotated by Ian Bruce.

page 827

**SCHOLION**

**814.** Haec praecepta pro motu corporum rigidorum determinando latissime patent, neque tantum ad motum liberum sunt adstricta; quomocunque enim eorum motus compescitur, sive super plano quodam sive iuxta alia corpora incedere cogantur, sive quodpiam eorum punctum fixum retineatur, quaestio semper ad tradita praecepta reduci potest. Scilicet qua parte aliud corpus contingunt, ibi dabitur pressio, quae primo indefinite in calculum introducta deinceps ita determinari debet, ut motus propositis conditionibus consentaneus reddatur; atque etiam hoc modo conflictus corporum explorabitur, Cuiusmodi investigationes antequam suscipiamus, casum quendam motus liberi expendi conveniet, in quo motus gyriorius circa axem variabilem locum inveniatur, dum corpus a viribus externis sollicitatur; cuiusmodi motus a vi gravitatis, quippe cuius directio per centrum inertiae cuiusque corporis transit, non producitur. Gravissima autem huius generis quaestio sine dubio in motu vertiginis corporum coelestium versatur, quae autem non nisi positis Astronomiae Theoreticae principiis suscipi potest. Consensu autem omnium observationum, quas adhuc instituere licuit, compertum est corpora coelestia perinde moveri, ac si se mutuo attraherent vel ad se invicem pellerentur viribus, quae sint in ratione reciproca duplicata distantiarum atque insuper massis proportionales. Scilicet quemadmodum quaevis corpora terram versus gravia sunt, ita etiam nisum quendam habent versus omnia corpora coelestia, qui eo maior evadat, quo magis quadratum distantiae diminuatur. Atque ex his viribus Astronomi motus progressivos corporum coelestium scrutari solent, quae investigatio cum ad motus punctorum sit referenda, hic tantum in motus gyriorios corporum coelestium inquiramus, quod argumentum in sequente capite generatim ita pertractare studebo, ut Astronomia inde haud contemnenda incrementa sit consecutura.